

Aviation Week

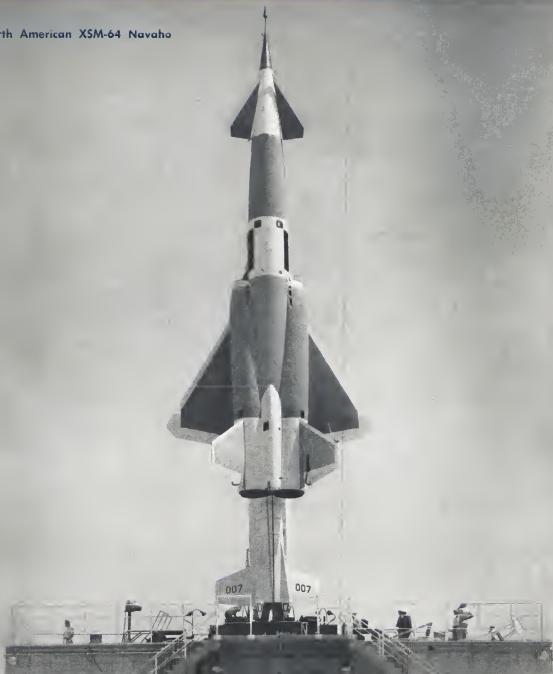
Including Space Technology

First Design
Analysis Of
Saab Draken

March 24, 1958 75 cents

A McGraw-Hill Publication

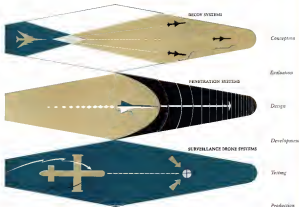
North American XSM-64 Navaho



DRONE and SMALL MISSILE SYSTEMS

Research and Development at Rheem Aircraft Division has a record of achievement in the field of drone and small missile systems.

The capability for complete "program management" is maintained in the line of current and completed projects and the areas of responsibility covered.



Rheem Aircraft is a division of world-wide Rheem Manufacturing Company which operates 17 plants in the United States; and with its associated and licensed companies operates 35 plants in 32 countries abroad. These extensive facilities coupled with Rheem's years of aircraft production experience provide the capability for the quantity production of drone and missile systems.

RHEEM MANUFACTURING CO./AIRCRAFT DIVISION
17711 WINDHILL AVENUE, DUBLIN, CALIFORNIA



INTER-OFFICE MEMORANDUM

To: Advertising Department Subject: Recruitment Advertising for Engineers
From: Vice President - Engineering

As the result of expanding activity in recent months, we have come up with a few openings for good engineers. Specifically, we need designers in the following product groups:

Pneumatic accessories
Electric motors
Fuel systems controls (pumps and valves)
Hydro-anti-skid braking system (electro-mechanical)

The requirements are not pressing; our present team can still carry the increased load. We're proud of the boys we now have, and it's important that we find exactly the right people to add to the group. Therefore, before you rush into print with the usual sort of "recruitment" advertising, here are some thoughts to use in formulating your message:

1. We don't hire engineers by the overload, and we don't stockpile them in reserve until the right project happens to come along. Our people are busy...very busy, and they like to be busy. They follow through on their ideas from inception to development and qualification. They feel a real responsibility for the hardware that finally results from their work. Therefore, we need more idea men with initiative and drive who are not afraid to get their hands dirty.
2. Don't write a lot of guff about "security" and "bright future." The kind of men we want carry their security around with them. They have the self-assurance that comes from ability and experience. Chances are they've known about our company for several years and have followed our progress in the industry. If they answer our ads, it means they like us - and they think they can help us to grow.
3. The kind of men we want will join us because we treat our engineers in the same way we treat our other key people. We don't isolate them and we don't put them on pedestals. We expect results; we know how to look for results; and we reward early when we find them.

One other point: the man who meets our requirements is probably too busy to write a long resume and application letter. Just tell him to call me personally, or to drop me a short note to let me know where I can contact him - to Hydro-Aire, Inc., 3030 Winona Avenue, Burbank, California. Phone: Victoria 9-1331.

Frank Cooper
Vice President -
Engineering & Sales

One of the two largest modules (right) prepared for the Superconducting Supercollider.

Transonic engine test No. 2 (center), with compressor and turbine area (left) showing in background.



Steel Platework by Pittsburgh-Des Moines



A MIGHTY INSTRUMENT OF DEFENSE

The Propulsion Wind Tunnel Facility at Tullahoma, Tennessee

The two largest wind tunnels of their kind in the country are shown built and under construction by PDM in the picture above. The Transonic Tunnel, now in operation, and the Mach 5 Supersonic Tunnel going up beside it each have interchangeable test sections 16' x 16' x 40' long—a measure of the great size of this massive project. The size and complexity of the work are in turn a measure of Pittsburgh-Des Moines' ability to meet dependably your most exacting platework requirements. Write us for engineering consultations and preliminary estimates on your forthcoming projects.



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Harrison smooths all corners—surface quality grades to the field of thermodynamics.

Quality qualifies Harrison for the rugged job of cooling Northrop guided missile

The Shark, sleek off in an arena of heat—has Harrison keep temperatures down to earth. With a transverse range, efficient heat control over the engine oil is a must for this spectacular platform bomber. And Harrison heat exchangers were selected to do the job.

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TEMPERATURES MADE TO ORDER
HARRISON

HARRISON RADIATOR DESIGN • GENERAL MOTORS CORPORATION • LANSING, MI 48106

New B. F. Goodrich Cladheat gives Convair 880 'hail-safe' de-icing protection

B.F. Goodrich



Entered in left wing apex and apex area



By full panel method 530 mph...



action of new B.F. Goodrich Cladheat continues to perform satisfactorily.

One of many safety features of the new Convair 880 commercial jet aircraft is the B.F. Goodrich Cladheat De-Ice system on the wingspan. Because the 880 will maintain unusually high cruising speeds in all kinds of weather, a de-ice system was needed that could withstand stresses from rain, dust—even hail—and still keep functioning dependably in spite of external damage.

The new B.F. Goodrich Cladheat De-Ice was selected because it meets this requirement, and also because it forms a smooth aerodynamic and low weight. Consisting of about one thousand heavy aluminum sandwiched between layers of semi-transparent glass fibers, the B.F. Goodrich De-Ice is installed in a single unit with a skin of anodized steel only .005 inches thick. This unit forms an integral part of the 880's empennage to give the de-icing system "hail safe" protection.

Every plane has its own special de-icing problem. And B.F. Goodrich has been solving these problems longer than any other company. For more information on Cladheat de-icing, send for the free booklet, "Electrotherm Products." Write B.F. Goodrich Aviation Products, a division of The B.F. Goodrich Company, Akron, Ohio.



B.F. Goodrich aviation products

EDITORIAL

Cape Canaveral Revisited

Last week we spent a gray, rainy day bridging instead the launching pads and blacksmiths of Florida's Cape Canaveral known as "range station No. 1" to the 14,000 people who now run the Air Force Missile Test Center. This was our third inspection of the Cape area, except for the palmetto scrub and the lighthouse, it would be hard to recognize it as the place we first saw each in 1953 when Richeson and Minister wander downed the test program and the Air Force's first missile testing group was firing Pencilball Larkie for practice.

Now the Cape is dominated by the six great test-towers and servicing gantries for the Convair Atlas and the Douglas Thor with construction work well along on four test launching complexes for the Martin Titan. We saw Atlas No. 3 on the pad going through preliminary checks for the most sophisticated test shot yet attempted with this intercontinental ballistic missile. On another pad, the Redstone short range ballistic missile of 1953 had become the first stage of Jupiter C scheduled to launch the third Explorer satellite later this week.

Divine Guidance

Near the tip of the Cape, the Northrop Snark is still being fed. In 1953, their early development project translated broken from commercial work about the "Snark related system of Cape Canaveral." Now the Snark are being 5,000 mi. downrange at high subsonic speeds with simulated navigation work and impacting near Ascension Island with guidance accuracy that top USAF development men are well not be achieved by any other missile for a long time.

At the new graph Vanguard pad there was an air of happy rehearsal and a feeling that the successful satellite shot on St. Patrick's day with a St. Christopher's medal added to the contractor's estimate proved that Divine Guidance is a real system requirement, not a matter of technical sophistication a facile human brain can contrive.

Perhaps the key point in attempting to evaluate the significance of what has happened at the Cape during the past three critical years of its growth is the fact that it has tentatively kept ahead of the missile state of the art so that when new types of missiles needed the development test stage the range facilities existed to handle them adequately. This is a tribute to Air Force research and development planners of a decade ago, who foresaw the need for this type of facility, as well as to the current agencies of the range and test center—Air Research and Development Command with its civilian contractors' Radio Corp. of America for range instrumentation and Pan American World Airways for range operation.

In a business where technical progress gallops at a rate that usually makes most test facilities obsolete before their concrete dries, this is a most unusual situation. When the Snark was ready to make its 5,000-mi. run

for accuracy, the range was sufficiently instrumented to handle it all the way to the Ascension impact area. Now that the Atlas and Thor are ready to make extreme altitude tests the Mark I Azusa tracking system is operational and capable of supplying pretty precise data on the missile's position and flight path. The Azusa system is another technical wonder! resulting from the old MX774 project at Convair in the mid-fifties that also produced the guiding rocket for control, integral test tracking and lightweight missile shell construction. The Azusa Mark I now operating at the Cape has also been combined with an IBM 704 computer to produce an impact predictor system that adds continuously to range safety by giving a running fix on where a missile will impact if it is destroyed at any time during its test flight.

When Atlas and Titan are ready to be tested for accuracy over their full 5,500 nautical mi. stretch, the range will be capable of handling them with a class of 11 telescopes and radar tracks extending over 100 miles beyond Ascension, an improved Mark II Azusa system installed at Eleuthera and an extremely long range radar at Trinidad.

This kind drive for effective range instrumentation has pushed the state of the art development head in both electronics and optics.

When the U.S. space program was finally formulated in 1955, the Cape was able to provide the place for it. It was able to get off the ground rail, even more important, the means to detect and progress from the inevitable early stage failures.

Even the ratlines of the road sign between the Cape and Patrick AFB have caught more of the significance of this facility planning. The construction of the new commercial Polaris road is just a shade behind the building program for the Navy's Polaris missile test complex.

Problem vs. Performance

The Missile Test Center and range now have assets totaling about 400 million taxpayer dollars. Nearly half of this sum has been poured into construction, range expansion, maintenance and new technical laboratory facilities since we last visited the Cape less than two years ago. Obviously a progress of this size and complexity equipped with such a tight time schedule is bound to generate a host of serious problems. It doesn't take an eagle reporter on the way back to assess a long list of gripes both test and management from disgruntled civilian and military personnel.

But based on years of watching the Cape grow from little more than a rattlesnake bar and scrub palmetto into an extremely accessible development test facility supporting the missile development programs of Air Force, Army and Navy, we believe looks are in order for the men of ARDC, Pan American and RCA who have planned, organized and executed this job.

—Robert Hutz

LME

GENERAL  ELECTRIC

LIGHT MILITARY ELECTRONIC EQUIPMENT DEPARTMENT
FRENCH ROAD, UTOCA, NEW YORK

In the Front Office

May Glen Kenneth P. Bragg, Air Research and Development Command, Deputy Commander for Air Defense Systems Integration and also head of the newly named Air Defense Systems Integration Division, USAF, Hirschman Field, Bedford, Mass.

Dr. D. W. Elms, a vice president, Helix Helicopters, Palo Alto, Calif. Dr. Elms will continue to head Adhesive Engineering Co. which is now a division of Helix.

Carl L. Sadley and Richard M. Olson, vice presidents, Southland Marine Trawl Co., Rockland, 12. Mr. Sadley will continue as general manager of Southland Trawler Division, Mr. Olson will be in charge of the new Southland Turbo Division.

For Goetz, a vice president, Waldbaum Frozen Imbusement Co., Waldbaum, Mass.
 Abrams & Deseris, a vice president
 Goetz Imbusement, Inc., Metuchen, N. J.,
 and general managers of the newly created
 Climate Imbusement Division.

George M. Daline, vice president and director of sales, Clontar Soap Service & Mfg. Co., Chicago, Ill.

Lt. Col. William J. Severs (USAF, ret.), assistant to the general manager, Hughes F-16B Division, Bell Aircraft Corp., 1000 N. Y.

Rear Adm. Frederic S. Widdington, from Chief of the Bureau of Ordnance, to Commander Naval Forces, Japan. **Rear Adm. Paul F. Beckwith**, from Chief of Staff and Aide, Commander, Naval Air Force, Atlantic Fleet, to Commander, U. S. Tenth Fleet.

Honors and Elections

Don L. Walter, vice president-engineering and Van Nostrand consultant for Vespacraft Aircraft Co., Inc. has been named Chairman of the National Advisory Committee for Aeronautics subcommittee on engine performance and operation, and a member of the committee on propellers for aircraft. Other NACA appointments include: **Charles C. Ross**, vice president-engineering for Aerquip Controls Corp., named chairman of the subcommittee on rocket engines, and a member of the committee on propellers for aircraft. **Dr. William J. O'Malley**, chief engineer for aircraft and missiles, directed work for Republic Aviation Corp., named chairman of the subcommittee on aircraft

Changes

W. F. Miller, manager of Gorman open stream-Camp Cooke Air Force Base, California; and Anthony, division of General Dynamics Corp., San Diego, Calif. E. W. Jurek also succeeds Mr. Miller as manager of Gorman operations-Edwards Rocket Base.

M Carl Hadden, director of marketing Caltrop, Denson, Lorchood, Aesch Corp., Baraback, Wis. Robert A. Bailey succeeds Mr. Hadden, as chief engineer, and Arthur E. Plank succeeds Mr. Bailey as chief advanced systems research engineer.

► Series of shaped modules for layman or high altitude tests is being built for potential industry users by Arnold Development Corp., Pasadena, Calif. (Cited in the market survey are varied to meet test requirements.)

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► Pratt & Whitney J56 turbojet being developed at the company's West Palm Beach facility is rated at about 10,000 lb. thrust and probably will be used in any Navy project utilizing nuclear power for aircraft propulsion.

► Use seven berthing and local hot spots as the rear backlogs of some hypersonic vehicles may prove as difficult to cool efficiently as the predictable maximum temperature zones on the leading edges. Hot spots would be caused by vortices which curl back over the backlogs as it develops left. Location of the hot spots would keep changing with angle of attack and speed.

► Air Force has set aside a proposal to equip the Lockheed F-104 with experimental cockpit controls similar to those being developed for the North American X-25 and other aircraft that attain altitudes where the atmosphere is not sufficiently dense for normal aerodynamic controls to function well.

►USAP's Range Aid Development Center has requested bids for installation in the Cockburn of prototype ORDR (orange/red digital radar) long-range radar return developed by Columbus University. Because the system determines azimuth by bifurcation, one transmitting and several receiving antennas located several hundred miles apart will be required. System will track missiles fired from Cape Canaveral for evaluation in possible eventual procurement for radar along the ballistic missile early warning line.

Each of three planned ballistic missile early warning sites to be installed for USAF by Radio Corp. of America will be operated by approximately 1,000 tech crew to be supplied by RCA Service Co. under a contract which will cost for two years after the sites become operational. In addition to Minuteman III type missiles under, each one will include two night beacon surveillance radar sweeping assigned horizontal angles to provide full coverage of Russia from the three sites (AW Feb. 24, p. 60).

* Manned ballistic rocket research system study is being participated in by McDonnell Aircraft, North American Aviation, Convair and Aero. Work may make use of Penn drug use studies.

✱Air Force has changed its operational requirements for Corvair's B-1B supersonic bomber, eliminating the photo-reconnaissance capability. As a result, Fairchild Camera has lost its contract for manufacture of the planned camera and associated equipment for this application. Cost of economic source pool program, including delivery of two prototypes, is \$18 million.

Rolls-Royce attempted to win China's Vaughn in General-purpose engines for the FB-117 all-weather fighter, but Chester Vaughn is sticking with the Pratt & Whitney J75. Rocket motor is the dual pulse engine. Chester Vaughn FB-117 will be supplied by Rocketdyne. The name of North American Aviation Co. lies in the choice of the Rocketdyne motor was ability to deliver a unit smaller than Rocketon Motors (AVF March 10, p. 21).

► New 5100 modules are to be installed on the Fiat G. 91 for tactical trials to meet a STAMPE requirement for ground attack. Single modules will be carried under each wing. New 5100 was developed as an air-to-air unit, weighs about 297 lb., is approximately 54 in. long (AW June 17, p. 58).

► Russia's Mi-6 twin turbine helicopter will carry a five-man crew—two pilots, an engineer, radio operator and armorer. Latter two operators are often omitted as short test hops. Chief designer M. I. Mi's top assistants in building the helicopter were leading engineers D. T. Metitskii and leading designer V. S. Odolenskiy.

■ Manquant's ramjet has been ground tested at 300,000 ft, and at a speed of more than Mach 2.5 in the Engine Test Facility at USAF's Arnold Engineering Development Center. Altitude is 30,000 ft, higher than that previously announced for air-breathing engines.

High Temperature Plastic

New plastic forms which enable molders to produce high temperature parts and feature that has been developed by Hughes Aircraft Co. Higher material will stand temperatures up to 5,000°F and retain rigidity at 5,000°F. Company expects it to be used in rocket nozzles, rocket blast defectors, jet nozzles, exhaust and nozzle shields for space-oriented ballistics missiles.

New plastic is an opaque polymer resin with random oriented reinforcing agent. Material has a self-healing feature under high temperature. A large portion of thermal energy is dissipated through random mixing of degradation products on the surface. Self-healing occurs when decomposition and accumulation does so much energy is lost from between model sheet and the molded material.

all liquid propellant vehicles with those interesting and important differences. At T-11 moments, the top cluster of solid propellant rockets and nozzle are not opening at 450 rpm. A generator lets the rocket motor slowly open about 20 sec. before cut-off of the first stage motor when rotation reaches its maximum 760 rpm.

Because of the generator is to keep the frequency of the opening rings from reaching a level where it would resonate with the frequency of the first stage. This opening of course, is required to stabilize the first stage and to prevent thrust vibration at cut-off one of the solid rockets fails to fire.

During countdown, a small package of dry ice is added to the vehicle so the top of the first stage is kept at the minimum amount of the circuit model and the T-13 around the motor, which is attached to long pole, is disconnected along with the "technical cut-off" of servicing lines which has been feeding into the engine.

At T-1 sec. 140 sec., two seconds in the first stage are closed. When they come a pressure drop in either the engine or fuel line this cut-off the engine.

The vehicle starts for two seconds after cut-off to make sure there is no residual thrust as the first stage which could start at its cutoff with the second stage after separation.

Explosion both leading the second stage to the first stage is detonated and no sparks really push the second stage ahead at an uncontrolled speed at the moment per second. In this point the Explorer is at an altitude of about 90 mi.

The vehicle continues to coast upward. At T-1 sec. 400 sec., the final three stages reach their apex. Two ground-controlled compressed air jets

align the final stage horizontally. The attitude control jets work in opposite directions of two so a proportional bias. As one outlet nozzle is closed by a certain amount, the opposite one is opened by the same amount. Two main control points, two jets. To get into orbit, Explorer 1 had to be within five degrees of horizontal, it was at 83 deg.

More difficult than obtaining horizontal alignment was the problem of determining and carrying the moment of apex. This is the point moment at which the final stages must be fired to bring the satellite to a velocity of about 6,500 mph to an orbital velocity of about 15,000 mph.

To determine this moment, project engineers used three different methods to predict apex: one based on data from a radar tracking, another, on the internal data from an accelerometer within the vehicle, the third, on data from ground-based Doppler signals. A sonogram was employed to get an average apex prediction and this was fed into a timer on the ground which started the rocket's auto signal. Actually, the auto signal was sent to the vehicle shortly before it reached apex in order to allow for burning time of the rocket.

Final burning time of the Explorer's 15 solid rockets was 15 sec. and this time had to be taken into account in calculating the moment of apex.

Data is still being received from the satellite's low power transmitter with one jet firing or the jet on signal. Signals from the high power transmitter ceased after 15 sec but inexplicably returned two after that. The rocket had not been fired. After 15 seconds the rocket level in the team's approach fell below a certain threshold required for signal generation, then, immediately like a "dead" car button, from a low moment despite a "star" button, they redoubled their effort, twice reaching a level above the critical threshold.

Dual-Thrust Motors

Dual-thrust solid propellant rocket motors are conceived as a practical alternative to the usual combination of separate booster and sustainer motors, and the dual thrust approach has been successfully applied to at least one case.

Dual thrust motor with a duration of over eight minutes has been successfully tested in the Tonno NKDT-1 target drive, and the dual thrust has been successfully applied to at least one case.

Comparison of the new approach with the separate booster and sustainer technique was made by George E. Miles of the Los Angeles Research and Phillips Petroleum Rocket Park Div.

son which developed the NKDT-1 motor.

Miles pointed out that integrating the booster and sustainer phases in a dual thrust motor eliminates the problems associated with jettisoning booster components. This means increased reliability, because there is no booster separation, and it eliminates the booster thrust alignment problem before launch.

Motor with a dual thrust motor could have the advantage of increased reliability due to character of space sequencing, could have a lower population motor gun weight, shorter length and lower population system cost. Such a propellant could improve chance for full mission control through cut flight because there is no booster to jettison, and, that, no change in acceleration configuration.

Miles concluded that the integrated dual thrust motor cuts the dual thrust of a spent booster phase while the sustainer is firing and that the dual thrust motor demands much closer and rather coordination of sustainer and propellant motor design. He said the choice between dual-thrust and separate motor often depends more on the specific requirements of a missile, than on comparative overall specific impulse.

Variety of configurations can be used with dual-thrust system. For thrust rates up to about 10, a single chamber unit upon preferable. Miles said in a single chamber unit, thrust level can be regulated by mechanically changing the nozzle throat area or by changing propellant grain geometry or composition.

Dual chamber unit has two separate chambers which can be arranged either concentrically or in tandem, according to Miles. This type of dual-thrust motor can use either single or multiple nozzles.

Practical application of the dual thrust approach in the NKDT-1 motor described by J. A. Cook, of Astrobleu. The motor gets its boost from a jet of higher burning rate propellant which is loaded to the full end of the sustainer grain, providing a simple, integrated two stage propellant system.

Sustainer phase propellant is a standard M15 grain type, and the boost phase is similar. Grain is a cylindrical cylindrical and burning design.

Motor is 10 in. dia and 59 in. long, plus a 20 in. blast tube which carries exhaust gas through the driver's tail nozzle. Motor is the center section, with nose and tail sections attached.

In the design the boost section of the motor burns for 3.5 sec., provides sea level thrust of 140 lb on a 500°F firing. Sustainer thrust at sea level is 10 lb, and the sustaining section burns 760 sec. The total thrust is 140 lb and it can operate up to 50,000 ft.



FROM which occurs Vanguard rocket angle of attack is adjusted by propellant injection. Agena-like driver is attached to top of Agena. Photo shows Agena. Note actual 325 ft. Agena sphere undergoing checks at left center of picture.

Space Technology

Vanguard's Success Forestalls Its Critics

By J. S. Bates, Jr.

Washington—Successful launch and orbit of Navy's Vanguard satellite last week, ended public protest which followed the agency's testing failure of the last three and a half months.

Disappointment and misunderstanding, which had taken a firm hold on large segments of Congress, the press and the public, seemed to pass and broad statements of faith in U.S. technology as the Navy Vanguard test vehicle sent a 325 lb sphere into orbit with a 2,515 mi. apogee.

"Though there is some feeling that the Vanguard is a marginal project that is moving behind schedule, the record indicates that it is one of the U.S.'s most successful high-performance rocket programs and has been progressing at an accelerated pace."

The project was initiated approximately two and one-half years ago, and the complete specification has been firm for five years than one and half years. This exceedingly short lead time makes the Vanguard one of the most advanced and efficient large U.S. rockets ever in flight test.

A number of improvements in structure and propulsion had been achieved.



CONTROLS of the Navy Vanguard TV-6 satellite were checked by Martin Co. engineers at Cape Canaveral, Fla., launching site. Rocket was supplied with a backup ground system to take control of the guidance located in the second stage if a failed



GANTRY is mated away from Vanguard in the prelaunching position approaching final position of third and second stages (not shown) using liquid oxygen as a charge from other vehicles in which the shot was fired.

have been made in the vehicle.

The project is well ahead of the original schedule which called for launching one fully instrumented 21.5 lb satellite before the end of the International Geophysical Year next Dec. 31. This plan was altered last July and the smaller 3.25 lb instrumented spheres were to be included in any complete test vehicle that would attempt to reach orbital speed. In this way, more useful information could be obtained from the preliminary tests.

The Vanguard satellite now in space is the result of one of these tests, which Navy officials have described as a near optimum performance by every rocket component. It was the first flight test for the second stage.

On the strength of this the schedule has been accelerated, and the next test vehicle will contain a 11.5 lb. satellite instead of the small sphere previously programmed. This enlarges the program from six to seven attempts to put fully instrumented spheres into orbit.

Account of the Vanguard landing has been in confused state ever since the White House announcement in the summer of 1955 which stated simply that the National Science Foundation would spend \$10 million on the satellite program. It has shifted through reports and queries that various Department of Defense funds would be added to bring this amount to a reasonable figure. Recent testimony before Congress indicates that total cost will stay below \$150 million, a modest amount for space development standards.

The Vanguard firing record is well above average for the early stages of a development program, and the project will be fortunate in outcome if through the seven launches which remain to be accomplished. Six test vehicles have been fired to date to prove out the rocket's various sections. Two of these flights have been successful for a 50% average, a very respectable figure. The first one the Soviet made satellite also shared the record, and the Vanguard was transformed from a military stepchild to a top priority program.

Technically, the rocket third test work had not made change on the previously announced design. It was equipped with a backup ground station in the control of the guidance located in the second stage if it failed. This would mean that the spin stabilized third stage would be launched at the proper angle and would enter at the necessary speed and would enter the ground station until radar tracking resumed and a computer to prevent rocket course information to an operator who could take control although this was not necessary last week.

The satellite itself was made of aluminum coated with a thin layer of silicon monoxide to control the ra-

VANGUARD instant launching which placed satellite sphere in orbit

dional temperature. This film reflects infrared, absorbing heat the specter depending upon this thickness. Two other beacon transmitters are carried in the sphere, one operating on batteries expected to last two weeks and the other from an solar cell in the outer radius. These were built by Vanguard by the Army and will provide localizing power in daylight as long as they are undamaged.

The orbital of one sphere in contact with the main structure and the other with the outer shell. Therefore, the absolute temperature of the two structures can be determined from the change in signal frequency.

The next Vanguard rocket, to be launched as soon as possible will contain a 21.5 lb. 30 in. sphere containing passively to measure ultra violet radiation in the wave length designated ionosphere. This type of ultra radiation affects radio communication and the weather. Measurements will also be made of ionospheric pressure and collisions with micrometeorites.

Later Vanguard satellites will carry on the strength of the earth's magnetic field, solar X-ray, cloud cover at the earth, total radiation of the sun, the reflected radiation of the earth and the density and distribution of cosmic rays.

Dr. Hagen is his latest statements indicated that Vanguard probably will not be selected after the conclusion of the current program. The complete vehicle or its various stages which are among the most efficient tests in satellite are being considered for near satellite space projects.

Competition Planned For USAF Minuteman

Garden City, N. Y.—An Arms & Ship competition for solid propellant intermediate and intermediate range ballistic missiles which ultimately will replace Atlas, Titan and Thor will be held during next six months according to Maj. Gen. Bernard A. Schriever. Figure is believed to be the Minuteman study package ballistic missile project (ASM Star 17, p. 21).

Commander of USAF's Ballistic Missile Division also revealed that Minuteman will employ an all manual guidance system, whose contractor also will be selected during next six months.

Costs below one million dollars during next year to Army division of Army's Ballistic Missile Corp. for ongoing at \$140 million contract covering development of several guidance systems for Titan, SCIM. Contract includes approximately \$75 million already spent by Army to date.





ENLARGED view of Soviet missile deployed in Moscow during the Nov. 7 anniversary parade may indicate experiments in reducing effective radar cross section, in the opinion of some experts, although otherwise seems would be to increase size of warhead as it does with some U.S. missiles.

Cross Section Threat to Missile Detection

By James A. Fiske

One potentially feasible way of reducing long range detection of missiles over more difficult is in reduction of the missile's effective radar cross section obtainable with techniques currently being explored in the U.S. and Russia.

This problem, and the cross problem, apparently have been postponed as "second priorities" in missions of effectiveness offered for Army's Nike-Zeus and Phalanx and USAF's Wenzel autonomous missile defense.

Normal radar for the radar cross section (aperture area) of an incoming (ICRM) without use of calculating range and probability of detection for these missile defense radar is 0.2 square meters, based on experimental work performed in 1958-59. Present technology, however, can reduce the apparent size of a missile by a factor of between 10 and 1,000.

Thereafter, the geometric shape with the smallest radar cross section when viewed from only one direction is an infinite cone area from the nose-on angle. The only backscattering of energy from such a shape would be from the tip of the cone, but the effective cross section in radar frequencies would be about one-half of a square meter.

For the practical problems of missile warheads, minimum cross section is obtained by approaching as much as possible the electrical characteristics of

the filament stage in adding a nose-shaped cone to the warhead and varying the contour of the filament base section to minimize backscattering and diffraction.

Reverse the cone shape it will achieve factors for re-entry, critical nose cone can be made shape that would be all-inclusive cone-shaped to provide good conductors, also as being studied.

With a conical nose, diffraction or backscattering of electrical energy around the rear surface of the warhead become the key problem. A hemispherical nose would diffuse energy around itself and direct it back in the direction of energy. A flat base would have sharp edges that would backscatter large amounts of energy.

The optimum shape, therefore, is a conical filament section, being into a base section that is curved but not a surface of revolution. That is, the rear section of the missile should look like a hemisphere angled so that energy diffracted around it will not return to the point of origin.

Also under study are techniques for avoiding spikes or sharp edges on the rear section capable of launching energy diffracted around the base in a rearward direction, and absorbing material to make those surfaces as nearly as possible to the cone in heat.

Both in the case of diffraction and backscattering there is a significant dependence on frequency.

• **Diffraction** The maximum energy

is returned in the direction of origin when the wave length approaches the order of the dimensions of the warhead, with a falloff toward zero in both directions from the maximum. In the area of maximum energy return, the energy level will oscillate rapidly with peaks and valleys as the warhead length changes.

• **Backscattering** Backscattering of energy from the cone tip and from any other point on the surface with wave length the larger the wavelength, the larger the energy return. The smaller the cone angle is, the less the energy that will be backscattered.

Studies of electromagnetic diffraction and scattering from bodies of different shapes indicate that backscattering, but some of the most important conclusions have been made by a Russian scientist, Vladimir Aleksandrovich Fok, who was awarded a Stalin

Correction

Due to a typographical error, Anthony Woods' March 17 story, page 18, is inadvertently reported that our pen was could mission for USAF's Minuteman multiple-purpose ballistic missile "is as pointed to be considerably higher than anything developed or proposed as far." The sentence should have read that our pen mission is expected to be "considerably lower than anything developed or proposed as far."

Peace, thus in 1946 for work looking up to a report entitled "Diffraction of Radio Waves Around the Earth's Surface."

Although the question is to whether Russia is ahead of this country in the area of study in regard to radar cross section, U.S. scientists must believe that the significance of Fok's work and its application to missile cross section reduction is not beginning to be considered here.

Much of Fok's papers on the subject have been translated and printed in a collection published last summer by Air Force Cambridge Research Center. Identification number of this document is AF-CRC-74-57-162.

Theoretical and experimental studies of the general scattering and diffraction problem are presently underway at the University of California at Berkeley, Ohio State University and the Institute of Mathematical Sciences of New York University.

Specific problems in reducing missile warhead cross section are being studied at the Air Force Cambridge Research Center, Cornell Aeronautical Laboratory, the University of Michigan, and Radio-technic, Inc.

A simple and direct method of analyzing problems of this type has been developed by Dr. Joseph Keller at SUTU Institute of Mathematical Sciences. This method is used to confirm more, especially in some respects to experimental evidence than the method of analysis used by Fok.

United Aircraft Names Two Vice Presidents

United Aircraft Corp. has announced two executive changes to become effective April 1. William A. Finkel will continue as chief of Plant & Wholesome Aircraft Division, will become United's vice president for engineering. Leonard G. Madril, new general manager of Plant & Wholesome's Connecticut operations, will become a vice president of the corporation and act as general manager of all of Plant & Wholesome's operations.

The changes will also reflect upon the succession of Leonard S. Babin as United vice chairman after 10 years with the corporation.

Stapp Reassigned

Dr. Col. John P. Stapp of aet last time has been assigned to Wright Air Development Center where he will head the USAF new Shocked Laboratory. Mr. David G. Stapp will also serve Stapp in Chief, Aeronautical Field Laboratory, Air Force Missile Development Center, Bellmore, NY.



Soviet Bisen Corries Nose Probe

New probe is made in the nose of Soviet fighter Bisen heavy bomber. Belongs on forward knowledge may enable enemies' counter radar detection and interception equipment.

Space Technology Gains Pose New Industry Fiscal Problems

Washington—Although Defense Department will plan \$1.8 billion in space procurement orders in the next month, period ending next June 30, Aircraft Industries Association President Gerald Cook warned last week that the rapid pace of technology has created financial problems of a magnitude that have not been seen in the aircraft industry.

Models and aircraft orders for the six months will total \$6.5 billion, compared to \$4.74 billion in the last half of fiscal 1958.

Aircraft orders for the January-June period will total \$4,652 million, or 57.2% of the fiscal 1958 obligation for aircraft. Model orders will total \$2,277 billion for the six months. That is 61.4% of total model obligation for the fiscal year.

Those two categories are reported to President Eisenhower in a letter from Defense Secretary Neil H. McElroy. Obligations for fiscal 1958 in current status total \$12 billion in January, McElroy said, but that will rise to an average of \$17 billion to \$18 billion for the last six months of fiscal 1958.

Military construction obligations, as much as in the last half of fiscal 1958, rose to \$37 million in January. The remaining \$17 billion in the program will be spent over the February-June period, McElroy said.

Advancements in attempting to use this increase in defense contracting to help the nation's economy is by using surplus labor force and small business.

Asst. Defense Secretary for Supply and Logistics Professor McGuire has asked the services to review construction orders and for more than \$1 million to give preference to surplus area and companies employing fewer than 500 persons whether companies are qualified and bids are no higher than bids from firms in non surplus areas.

McElroy also asked Congress to extend the Researcher's Act for two years. His projected expenditures subject to appropriations for fiscal 1957 at \$18 billion and estimated this week are \$19 billion each for fiscal 1958 and 1959.

Many improvements in pricing policies and contracts in technology have been achieved, McElroy said, but developing technology and increasing complexity of weapons have made a harder task on the present the possibility of even greater expansion. Board now is studying how to draw outside experts behind in its determination of profits.

McElroy's investigation reflects concern in the sharp criticism from AIA, Post, and Congress, along with the industry's loss, coming late, decreased payoffs, personnel, increased responsibilities for increasing research development for new production facilities and increased difficulty in acquiring capital to finance it.

Cost, increased the lack of demand in government, and said the tempo of the American people is such that they can expect to receive adequate

lynds, whenever Douglas is asked that they are needed."

Industry's problems, he said, are financial. Cook noted the use of 15 major airlines, missile and engine manufacturers who he said:

- **Successful** not investment in facilities for \$10 million to \$40 million from December, 1958, to December, 1959—an additional investment of \$180 million after interest and depreciation. These facilities are generally used for development and production of military weapons, Cook said. The money "rotted"—and if the type of weapons that will ultimately be required in 1960-65 changes significantly, (a) could a lot then be described as "fossilized," he said.

- **Many existing** backlog can be wiped out now suddenly if are new profitable developments are made that make obsolete a particular product under manufacture."

Industry's investment in acquisition receivable and inventory from \$701 million to \$2,152 billion in the case six years, partly as a result of progressively reduced program payments for work already performed on fixed-price and cost-plus-fixed-fee contracts.

Reduced program payments have forced the industry to finance a larger share of work for government than for

commercial customers, Cook said.

Three principal ways to acquire capital and their relation to industry problems, Cook said, are:

- **Retained earnings.** This industry has for a number of years generated a greater percentage of its earnings than any other manufacturing industry.

- **Capital** that can be accumulated this year is limited by total earnings and by stockholder decisions, "partially, if one ever hopes to compete in the space market again for new equity capital."

- **Borrowings.** "From Dec. 31, 1959, through Dec. 31, 1957, 15 of our major airlines, missile and engine manufacturers increased their borrowings from \$25 million to \$561 million, or 27 fold," Cook said. "We certainly are approaching the limit of our borrowing capacity."

- **Sale of equity securities.** Cook quoted a recent report of the Aviation Security Committee of the Investment Business Association of America, which said that "three competitors in the aircraft market have reduced aircraft manufacturers to a low priority for new equity securities." Due to the dollar shift, Defense Department policy as much as to insure industry problems, the investment community has judged aircraft manufacturers' utility inadequate for the risk involved."

Norden-Kelley Bid

East Hartford, Conn.—United Aircraft Corp. has offered to buy the assets and business of Norden-Kelley Corp., and Norden-Kelley's decision has appeared. The offer is subject to approval of United Aircraft and approval of Norden-Kelley stockholders. United would exchange one share of common stock for 10 shares of the newer firm.

a record except for World War II earnings. Net income dropped from \$10,600,000 in 1956 to \$9,000,000, a result of the Defense Department cost containment changes.

- **Chrysler Aircraft** Inc., doubled its sales over 1958, reaching record for the company at \$277,752,770. Net income was \$6,152,451, or \$5.65 per common share, compared with \$4,155,161, or \$3.81 per common share. Grouping declined its operations since 1958, pointing out its backlog of \$670 million in aircraft and missile orders versus substantial production in 1960.

- **Chrysler Corp.** Defense order backlog held by Chrysler Corp. increased last year from \$100 million to \$309 million. This includes medium tanks, fire control equipment and trucks in addition to its contracts for Army Jeeps and Redstone missiles and a utility VTOL vehicle. Chrysler also posted a new high of \$1.54 billion with defense sales contributing 1.5% to \$275 million.

- **Thompson Products.** With 72% of its operations devoted to defense products, Thompson Products reported 1957 sales of \$165,578,425. Net income of \$11,343,014, or \$4.39 a share. Its major industrial contracts from \$13,612,685 to \$4,610 a share in 1956.

- **Kaiser Aircraft.** Record sales of \$15,195,760, compared with \$12,364,555 the year before, were reported by Kaiser Aircraft Corp.

UAW Signs Contract With Chance Vought

Dallas—UAW contract ultimatum in the current round of labor negotiations with major aircraft companies came last week when Chance Vought Aircraft Inc. signed a contract with its United Auto Workers Union local.

Economic adjustment sales up a package of about 14 cents an hour, including wage increases and fringe benefits. This falls remarkably short of the package called for by the union, which is now being vigorously demanded by the UAW local. Wage increases range from eight to 14 cents an hour.

Union accepted a two-year contract with a clause that allows renegotiating of



WV-2E Carries 9-Ton Rotadome

New York Rotadome and support crane by Lockheed's WV-2E rotor Super Constellation, delivered last week to the Navy Air Development Unit, North Westport, Mass., where it will undergo its aerial tests and evaluation. New AN/APN-58 airborne radar warning radar mounted with 37 ft. diameter rotadome mounted vertically with the rotating radome will provide approximately 7000 sq. mi. range from AN/APN-58 radar of standard WV-2E aircraft. Eventually Navy hopes to add height-finding capability to AN/APN-58, but WV-2E presently is equipped with AN/APN-58 height-finding radar with rotating radome mounted in Rotadome support yoke. The rotadome rotates around this with 3700 lb. of load in forward baggage compartment, but each of the rotadome will be replaced by additional electronic equipment as it is delivered by the manufacturer.

the contract on wages and contract provisions at the end of the first year.

While the union didn't get the cost of living escalator clause, it wanted Chance Vought did agree to improve the automatic progression portion of its wage plan, providing raises to about 5000 employees in addition to the pay (a) is up increases.

Company accepted the principle of corporate contribution to selected areas but held firm against UAW demand for automatic Union share cost reduction. The right to make a grievance, and grievance procedures were also shared.

Provisions were included in the new contract in one aspect during by which and the company agreed to extend union representation to employees transferred to off-site base and will continue the union dues check-off for each employee.

Maj. Simons to Receive Two FAI Awards

Los Angeles—Air Force Maj. Daniel C. Simons will be awarded the Gold Medal of the Aerospace Industries International at FAI's 51st General Conference here on April 14 for his

three billion flight to 101,516 ft.

Maj. Simons also will receive FAI's de Louven Medal, which is awarded to persons who have broken world absolute records for his balloon flight last Aug. 19-20.

The annual Gold Medal award is presented "to those persons who have contributed greatly to the development of aeronautics by their active work, achievement, initiative or devotion to the cause of aviation."

The de Louven Medal also will be given to USAF Maj. Adrian D. Dora who established the world speed record of 1,330 ft. high, last December on a McDonnell F-101.

The de Louven Diplomas will be presented to three US recipients at the conference for "their skill, initiative and devotion" in serving the cause of aviation.

They are:

- **Robert R. Hake**, editor of *Aviation Week*, for extensive aviation publicity during his assignment as the American public on the progress and development of U.S. aviation.

- **Joe Cline**, president of the Pan American Club of America, "for the effort expended in connection with the program and development of the spirit of journalism in the U.S."

- **Dr. Leslie A. Brown**, director of the Institute of Aviation, University of Illinois, "for the effort expended in connection with the program and development of aviation education in the U.S."
- **Robert L. Lewis**, editor of the *Aircraft Models*, "for the effort expended in connection with the program and development of aeronautical activities in the U.S."

News Digest

Dr. Gen Donald L. Pett, Air Force deputy chief of staff for development will leave this weekend shortly after completion of 30 years of service on May 20. He will be succeeded by Maj. Gen. Robert C. Wilson, new USAF member of Defense Department's Weapons System Evaluation Group. Gen. Pett is expected to take a post in private industry. In another move, Lt. Col. William H. Turner, deputy chief of staff for operations, will succeed Irving L. Cox, Joseph Smith as commander of the Military Air Support Service. Lt. Col. David C. Stroup, commander of the Air University at Maxwell, Ala., will succeed Gen. Turner at the Pentagon.

Borrowing Marks Industry Reports

New York—Douglas Aircraft Co.'s plan to sell up to \$1 billion in new convertible sinking fund debentures is regarded by financial observers here as the last borrowing step Douglas will need to finance the development expenses of its DC-12 jet transport.

Douglas' strategy is complicated by reduction by the government in program payments. Previously a \$30 million loss of bank credit had been established by the company, and next last year it sold \$27 million in new convertible debentures.

With the proposed new issue, Douglas would have been working nearly available for DC-12. The company plan relies on its own inventory funds, a draw on the company by the government payment reduction.

Financial reports of other aviation companies reflected this same problem last week. Thompson Products, Inc., sale of stock subsidiaries it was forced to make in the wake of Defense Department cuts.

Security of these funds allegedly is the reason the firm launched the last Special, the Thompson report said. Incoming orders since the first of 1959 have been higher than the last months of 1957, but this may not indicate results for the new Thompson and

J57 Overhaul

East Hartford, Conn.—USAF has extended the time between overhauls for the Pratt & Whitney J57 engine from 800 to 1,000 hours. The J57 now has seven or 1075 hours in 1971 with overhaul time of 10 hours. Last Douglas has 1968 Bombardier Wing of the Stinson Air Command reported that J57s in its fleet at its Western Air Base, near Colorado Springs, after flight hours without an engine have and all the time for repair or overhaul.

Soviet Stand May Block Aeroflot Growth

Russia's refusal to reciprocal exchanges could bar New York route; threatens British bilateral pact.

By L. L. Doty

Washington—Russia's plans to expand its international air routes outside the Iron Curtain have hit an obstacle that may bar the way to direct air service between Moscow and New York.

Russian refusal to exchange air rights on a reciprocal basis already threatens to delay implementation of service between Moscow and London under the terms of a bilateral agreement signed with Britain late last year (AW Dec 10, p. 31). Service between the two cities will be scheduled to begin in June or July.

Aeroflot Problems

That are some of the factors blocking Russia's expansion program for Aeroflot, Soviet-owned airline.

- **Russian** have historically avoided sea-fall in bilateral negotiations of fifth freedom rights, the right of one country to carry traffic from a second country to a third. U.S. will meet upon the admission of fifth freedom in air transport agreements signed with Russia.

- **Negotiations between Russia** and Scandinavian countries covering transit privileges in the right of Aeroflot to fly beyond Copenhagen to London were suspended early this month. The Scandinavian countries are backed by the U.S. in their stand that such rights will be granted only in exchange for similar rights to Scandinavian Airlines System (SAS) to fly beyond Moscow.

- **Soviet Union** is not a member of either the International Civil Aviation Organization or the International Air Transport Association. U.S. will stand firm on both IATA standards and ICAO standards, including airworthiness and the use of the English language as an international language.

- **British** have considered to the use of the Russian language in the Moscow airport, English at the London airport.
- **National level of the T-104** jet transport, scheduled for service by Aeroflot on the London-Moscow route, fails to meet British requirements. A high nose level will certainly hamper loading equipment from New York International Airport.

When the bilateral agreement between Britain and the Soviet Union for the London-Moscow route was signed last year, it was no obstacle expected to stand in the way of negotiating service

between the two countries for the first time. The Soviets made a number of major concessions in order to reach a first agreement.

Russia, however, has since an agreement which is in place when it opened to the British the Scandinavian companies on transit and landing rights in Copenhagen. Although Aeroflot now holds "transit" rights in Copenhagen under the terms of a bilateral agreement signed March 11, 1956 for direct service between Copenhagen and Moscow, it does not have rights to fly beyond the Danish capital.

It is the first time that the Soviet Union has been faced with the problem of negotiating for access within Soviet territory other than the direct Moscow route. With one exception, the recently signed agreement between India and Russia-Soviet bilateral agreements almost have been on a straight transit-to-terminal basis. The Soviets have never touched on fifth freedom rights.

Soviet Concessions

In the case of the bilateral agreement with India, Moscow was forced to make a number of concessions to the Afghan government in order to obtain transit rights through Kabul. In the agreement, Aeroflot gained rights to fly through the Afghan capital to New Delhi in exchange for rights to the Afghan airport to fly beyond the city of Moscow.

The agreement does require that only such pilots serve as crew members and that Afghan pilots, suitable aircraft or even an airline, the Russian offer of Moscow transit rights are hardly onerous.

A demand by the Indian government for a route beyond Moscow in exchange for an Aeroflot route beyond New Delhi was fully reported in the Russian press. It seems to appear in the Soviet press that Russia is open to talks in Russia and Indonesia. In addition, the Russians indicated India could make good its intent to fly a commercial route through Moscow to West Asia.

Russian attempts to bargain with Japan for a route into Tokyo are apt likely to show some progress unless the Japanese are given an opportunity to negotiate with Moscow. One offer by Communist Paris, Secretary Nikita Khrushchev for an exchange of routes

between Tokyo and Khabarovsk was promptly turned down by the Japanese.

Japan wants a direct route to Moscow to an important link to Western Europe and probably will hold out for such a route before it goes on to Russia.

'Beyond' Moscow Request

SAS wants beyond Moscow rights in a first step toward another polar route to the Far East. One study plan is the mid-air, stop-gate plan between London and Tokyo at 5,500 ft as compared to the distance airframe are forced to fly as far as the Soviet Union is in both halves of Iron Curtain nations.

How far the Soviet Union is in making bilateral concessions in order to fulfill its long range program is still a matter of conjecture. During the Stalin regime, no commercial flights operated by either Communist or western nations were allowed within Russia proper.

In early 1955 agreements were made with a number of satellite countries in a first step in the development of Soviet commercial aviation. First agreement with a nation outside the Communist sphere of influence was with Sweden.

It now appears evident that Russia has no intention of being left behind in its transportation program and, build up by a significant fleet of modern jet transport aircraft in making a serious bid for a prestige ranking in commercial aviation.

1950 Objectives

Completion of bilateral agreements with India and the United Kingdom were lower priority than American in major 1950 objectives in Aeroflot's expansion. Air France in cooperation with Aeroflot on an interline agreement pending for direct connections between the two airlines for Bagdad, on the Paris-Moscow route. However, negotiations between France and Russia covering bilateral agreements have failed so far at least two occasions.

The Soviets will begin talks with the U.S. on a bilateral agreement this fall. First talking that Russia was interested in opening negotiations arose in part of a cultural exchange agreement designed to improve mutual understanding (AW Feb. 5, p. 47). The Soviet Union and the U.S. agreed to principle on the establishment of reciprocal direct air rights between the U.S. and the Soviet Union.

Most observers feel that Russia has toward the talks to conclude with the availability of the language T-104

transporters (AW Feb. 17, p. 31). In addition, it is believed that the Soviets want to develop a transit experience of about six months with the British agreement in a form for bargaining with the U.S.

The British agreement with the Soviet Union does not differ essentially from that signed with Sweden. The picture is substantially the same.

It is Soviet practice to negotiate directly with the government instead of as principals of the agreement. All nations pertaining to technical requirements and operational problems are then negotiated between the two air lines designated for the route or routes. Aeroflot, however, is a part of the government. Full title of the airline is the Civil Air Fleet of the Council of Ministers of the USSR. Head of the organization's control board is Chief Air Marshal Pavel Zhigalov.

Negotiations for the U.S. will be

conducted by the State Department as usual. The American World Agency, the only U.S. carrier certified to operate into Moscow, taking an active part.

First Stand

In recent months, the State Department has taken a firm stand on U.S. rights in negotiations with other countries. The stand has perhaps offset earlier industry changes that State has been actively concerned in granting traffic rights to foreign carriers. In addition, the State Department has indicated that it will make no major decisions from standard practices in dealing with the Russians and will work closely with the U.S. current basic working air-fuel document.

Industry officials probably will show more concern over airframe and engine standards than on air-sea law in dealing with the Russian. Latest

ICAO code says not a broad but abstract set of standards for airframe, engine and then permits member nations to establish national standards within the code as a basis for certification of aircraft.

As a result, such member nations as a position to implement their will the airframe, engine, and air-craft operated by other member nations. Some U.S. airline officials fear that there will be no method of ensuring the safety of Soviet aircraft so long as the country remains outside of ICAO membership even though there are no inspection systems within ICAO to police airframe.

The British agreement does state that standards presented and established or recommended by ICAO shall be adopted. It also requires that aircraft shall be equipped to use ILS and one or more navigation aids with appropriate radio frequencies for use



Plane to Carry Armed SAC Crew Rescue Vehicle

New type of short field takeoff and landing aircraft required to execute the Air Force rescue and recovery mission in downed aircraft. Plane is designed to carry Strategic Air Command bomber crew who are forced to land out over enemy territory. Currently USAF has only C-47s and G-119s for this mission. New design concept is for a turboprop powered aircraft with short field takeoff and high landing gear for use on rough or unpaved fields. It would descend lightweight mobile weapons carrier as loaded in place. Vehicle can be loaded or unloaded without the use of the main cabin door. Helicopters are not regarded as practical for this operation because of the relatively long distances made during recovery from which SAC crew men have to be rescued.





OUTAWAY model of 72-seat East German BB-152 jet transport reveals cabin and flight deck seating arrangement, refuel decending scheme.

East Germans Display BB-152 Mockups

Layoffs—first official details of the East German BB-152 jet transport and its Type 014 powerplants have been shown here in the Technical Section of the annual Leipzig Trade Fair.

A turbo-fan model of the transport aircraft, designed by Rostock German Poly. Brunell Bunde, was shown together with a full-size mockup at model of the engine (AW May 17, p. 36).

The 72-seat transport prototype is scheduled to fly in May and to enter service with the East German Deutsche Luftflotte in 1968. Proposed routes

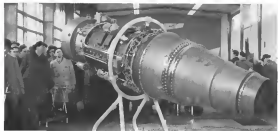
will be Berlin-Moscow and Berlin-Sofia and being flown by Russian designed Ilyushin 11B.

Gross weight of the transport is 102,000 lb., and its payload is given as about 12,000 lb. Maximum stage length is 1,660 m.

Model shows a conventional wing wing layout with five engines mounted in paired pods. Branch-ductors (fans) are positioned on either side of the pod. Shape of the rear portion of the nacelle includes a bulge for housing the main landing gear bogies. Engines ex-

haust on either side of the housing. Bunde's design does not have two refined fans, and the quoted cruise speed of 530 to 550 mph seems optimistic in view of the relatively short fuselage and the moderate wing sweep and high thickness-chord ratio.

Cabin interior is decorated with pale blue ceilings and walls, darker blue chair with mustard fabric and deep blue rug. Window curtains are a light brown. Varnished wood windows appear in one view of the mockup with a hand-grip at the top for pulling out the en-



TYPE 014 powerplant mockup has 12-stage axial compressor, turbine combustion chamber with 12 individual burners, two-stage turbine.

ter, and for emergency escape.

Seating appears to be five-above-five below, with five seats per row on one side of the aisle. Flight deck shows position for a radio operator and possibly a flight engineer, in addition to the pilot and copilot.

Wing span of the BB-152 is 58 ft 11 in., overall length is 103 ft, and height is just under 30 ft. Wing area is 1,470 sq ft.

Takeoff run is given as 3,180 ft, landing speed is 120 mph. Cruise is maintained to 6,400 ft at cruise altitude.

Plan of the Type 014 powerplant is 8,550 lb. The engine has a 12-stage axial compressor, a ramjet combustion chamber with 12 individual burners and a two-stage turbine. Officers with long antennae surrounded the German wartime fighters from tests of jet engines, which included a Type 014 project of similar description.

Fuel consumption of the engine, presumably under cruise conditions, is given as 0.15 lb./hr. per lb. of thrust. Fuel flow is 130 lb./hr. per lb. of thrust. Engine overall length is 160 ft 11 in., and diameter is 35.5 in. Both engine and nacelle are constructed in metallic products without fan case (nacelle) and the nacelle is not to be used in takeoff. This is true of the design specifications of the latest Russian developments showing in the East German products.

East German's aircraft industry (Volkswagen Luftfahrtindustrie der Deutschen Demokratischen Republik) took root in the People's German Democratic Republic in the GDR. The aircraft industry in the GDR (German Democratic Republic) founded out just about four years ago and started to produce. Dornier 114 jet engines (transport model) have been in the inventory at Dresden, a reasonable new building the BB-152 prototypes.

Communication Plans Requested by AMB

Washington—Aerovis Mediterranean Board has asked for studies, proposals by April 30 for development of a two-way automatic general air-ground communication system (AGMS), or data link, to be used for operational flight tests by Feb. 1, 1969.

Specifications for the new data link, which will enable traffic coordination and plan to exchange routine messages by push button to relieve overloaded radio communications were released to representatives of 42 aviation companies who attended recent AMB meeting here. Indicators are that a sizable number of the firms will submit proposals.

Specifications call for a data link system operating in the very high frequency (VHF) band now used for civil

AIRLINE OBSERVER

U.S. airlines are showing increasing concern over frequency of flights scheduled this summer on North Atlantic routes and will ask the State Department to review such reports expected to be issued. The U.S. industry feels that much of the traffic originating in New York on west coast routes is KLM, SAS and Sabena is fifth European traffic and a number, estimated by the Bermuda agreement. Pan Am services can be expected to argue that the traffic is not fifth because even though it is transported from the U.S. to a point beyond the terminal of the carrier involved. The terminal stop they will see as only a stopover point at a non-terminal station and not a pickup point.

AirlineGeneral is expected to begin flight tests shortly on its reduced passenger seating indicator fitted out with new optics. General tests will compare optical indicators that it can cancel with indication of photo-type models for more accurate evaluation.

Northeast Airlines is back in the market for turbine-powered transports but has not gone beyond the looking stage. Northeast officials are watching business performance closely on British Overseas Airways Corp. and El Al Airlines. Western Airlines is planning a strong bid to Northeast in an attempt to avoid some interest in the T-7. Western's approach could call for El Al Airlines. Meanwhile, British Overseas Corp. E. Gardner and President James W. Austin have been on the West Coast in prospective treatment for a fleet of American-made turbine transport.

Lake Air Transport announced survey of seasonal passengers indicates that seasonal demand for about 60% of total traffic is compared with between 13 and 14% for non-seasonal. Actual reduction of no-shows since the last time of the three part control plan was reported in Sept. 1966 now amounts to 35.5%. According to the report, airlines are meeting about 22% of the seasonal demand within four hours of flight time.

Mr. Marshall P. F. Zagari, head of Boeing's Aircraft Division, says the Boeing plant will begin to take delivery on a large number of 747 and 727 aircraft. The plant will begin to take delivery on a large number of 747 and 727 aircraft. The plant will begin to take delivery on a large number of 747 and 727 aircraft.

Dorval Air Transport, one of the largest of Canadian air freight operators, has been placed into receivership following a petition for bankruptcy filed by Imperial Oil Ltd. of Toronto.

Tension still between Japan and the U.S. on a revision of the bilateral agreement between the two countries will begin in mid-April. Japanese delegation is in Washington now discussing preliminary subjects pertaining to the agreement.

Civil Aeronautics Board has approved the new "economy" service on North Atlantic routes to be effective April 1. The new rate of \$232 for a flight between New York and London is \$35 less than present tourist fares. Board also approved discontinuance of the present 15-day round trip excursion fare.

Thrust World Airlines has sold a fleet of six DC-8 to Eastern Aircraft Corp. of Blackhawk, N. J. Martin Ferraro, co-owner of Eastern Airlines, said he plans to use the aircraft, which were flown in cargo planes by TWA, on a newly organized airline called Aerostar which will operate on Gomara. Ferraro is president of Aerostar.

Initial runway configuration proposed for Charlotte Airport, aimed at part for Washington, is showing deep pockets from which pilots. They charge that planned northwesterly parallel runways are not suitable for present land conditions in the area. Pilots need deep runways set of northwest-southeast and northwest-southeast angles, a configuration that will require the purchase of additional acreage for the airport. Pilots also warn that land to be purchased under proposed runway arrangement is not sufficient for future expansion.



De Havilland 121 Jetliner Selected by BEA

British European Airways plans to order 21 de Havilland 121 jetliners, with options for 12 more. Airbus, powered by three Rolls-Royce RB. 548 engines, can hold 70-80 passengers in standard seating configuration up to 300 in high density seat plan. Maximum cruising speed will be more than 600 mph. Plane is designed to operate from 6,000-ft. runways. Engines will develop 12,000 lb. thrust.

voice communications. System is to use existing VHF communications receivers and transmitters designed to Aeronautical Radio Inc. characteristic 120A and also is to be able to operate with military ultra high frequency (UHF) receivers and transmitters.

System proposed by AMB is to be able to handle up to 100 aircraft (flying within line-of-sight range of station) and provide interchange of information at least once every two seconds.

Through ground data link stations, traffic controller or traffic control data station will be able to transmit two types of messages addressed to a specific aircraft.

• Class of up to 12 different routine messages, such as request for repair, proposal, waste contact requested, change over to new frequency, emergency alarm.

• Class of up to 12 different messages conveying specific information, such as proceed to fix, "OK" held at "N" alt hold.

Plan's advanced data link also will enable him to transmit both types of messages, providing a choice of one of up to 12 routine messages or 12 specific information messages, the latter going airplane portions, altitude and track over the fix.

AMB is asking that system provide, if possible, one million different "ad-

dresses" to permit a specific one to be assigned to each aircraft and used with out change for its lifetime.

The agency says it also would be desirable if the system permitted an emergency message to be automatically transmitted to all aircraft in the area without filling time for a "call call."

Aerovox Modernization Board specifications calls for use of techniques to assure that a message intended for one aircraft cannot be accidentally received by another. Specification also calls for airborne equipment to automatically alert the pilot whenever it fails to receive a ground-station interrogation for a period of more than five minutes.

to include dispatchers at each point as Leningrad, Moscow, Kharkov, Odessa, Yalta, Sochi on the Black Sea and Yelen in the Caucasus. The team will be flown an Air France to Prague with representatives there on Aeroflot's new Tupolev jet aircraft to enter in the USSR.

• **Continental Air Lines**, recently recommended by Civil Aeronautics Board for new service on routes between the East and the West Coast says it will serve these routes with Boeing 707 turboprops, turboprop Vickers Viscount II Viscounts and Douglas DC 7Bs.

• **Soviet** carried 932,911 overseas passengers on its world network during 1957, an increase of 28% over the 1956 figure. The airline carried 13,611 tons of freight and 4,796 tons of mail, with a total of 114,898,610 ton-kilometers flown. The airline had losses of \$1.95.

• **Trans-Pacific (Aloha) Airlines** reports a net profit of \$64,771 for 1957 despite an operating loss of \$79,815. The loss was offset by revenues of \$110,668 from the sale of one DC-3 and \$10,665 from military contracts. The airline's passenger volume increased from 270,100 in 1956 to 223,858 in 1957.



Lockheed's C-47B "Beechcraft" turboprop transport will have Hamilton Standard & Hydromatic propellers. They maintain it another example of Hamilton Standard's leadership in the design and production of propellers, propellers or electronic components for more than 50 modern types of turbine or piston powered aircraft in service.

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SHORTLINES

• **Air Algérie** has ordered two Sud Aviation four-engine Caravelle transports with delivery scheduled to early 1968. The Algerian carrier operates routes between Algiers and France and Switzerland as well as within Algeria. The Air Algérie order brings to 25 the number of Caravelles sold to date with options for another 15.

• **Air France** plans three orders of the Soviet Union for 20, 25 and 32 jets

EA

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SPACE TECHNOLOGY



NAVY says man could survive on moon in 22-ft. full-pressure suit (left) designed for Navy by E. F. Goodrich Co. It has now been replaced by lighter fabric suit. Right, North American Aviation test pilot Scott Crossfield pilots X-15 mockup in garb of Navy coveralls.

Navy Details Needs for Space Mission

By Cecil Baseler

Washington—Navy believes it needs to go to space with both manned and unmanned vehicles to do an effective job in the sea. It hopes to go there as part of a national program involving all three military services.

Freedom and rapid maneuverability were in which Navy says it has space superiority are in the fields of reconnaissance, communications relay and defense against.

Navy officials also foresee the possibility of using satellites in anti-submarine warfare, monitoring submarines, ocean waves, and as platforms for mapping ships and aircraft movements.

Because of Aerotherm already has conducted a hypothetical study named "torpedo," placing a hypersonic glider aboard a submarine in a submarine follow-up to the Polaris fleet ballistic missile weapons system and with roughly the same advantages and problems. One of these problems, Navy officials be-

lieve, could be effectively solved by a navigation satellite.

"Artificial star" satellites they say could give the commanders of a Polaris submarine a precise fix on the location of his vessel in relation to its intended target.

They also believe advanced satellites might be used to relay a firing order to the submarine commander and then, on the next orbit, tell him the result of his shot.

Such a satellite would require a naviga-



CROSSFIELD (left) "flies" coveralls garb. High G forces led to right hand muscle stick atrophies. At right, pilot in coveralls garb sits under force of his Gs.

wide tracking system to provide coast and complete data on the ship, which Navy said it could provide through a network of mobile telemetry ships.

The service already is passing valuable space information from its earth-south Missouri system facilities to make the missile's radar find its target, with the International Geophysical Year and plan a similar range to extend across the Pacific from the National Pacific Missile Test Range at Pt. Mugu, Calif. (AW March 17, p. 21).

These tracking stations will give an experience with the appearance of reconnaissance satellites and trained or automated space vehicles designed to receive the earth's atmosphere.

The Missouri stations will monitor the reconnaissance satellites, collect and reduce the data transmitted from them. In modified form, they also can be used to transmit effective control over a manned or unmanned re-entry vehicle, lengthening the orbit in controlled steps to gradually slow the vehicle and make re-entry possible and guiding against any human control failure.

X-15 Research

North American's X-15 high-altitude research vehicle, for example, will be launched on its initial flight early next year by Navy's island tracking range extending from Pt. Mugu to Dupont, Utah, 900 mi away. The X-15 will take off from the far end of the range at Dupont and follow a westward course to Edwards AFB, Calif. The island range will follow its flight reduce data on its performance and exercise some control over the aircraft in the event of serious pilot error.

Navy also has been active since 1963 in attempting to solve the human factor factors in space flight and is deeply involved in this aspect of the X-15 program.

Working with the National Advisory Committee for Aeronautics, Navy has signed the large contract at its Ames Research Laboratory, Moffett, Calif., to accurately simulate the flight conditions the high-altitude flight will encounter. The recordings, believed to be the world's largest, has been modified to provide an exact duplication of the X-15's cockpit control placement, manual and visual, and has been extensively flown by the first three men scheduled to pilot the earth-North American Aviation test pilot Scott Crossfield, Air Force Captain Kenneth A. Collins and NASA test pilot Joseph Walker.

In planning for the program to study pilot capabilities and to find means of compensating for human shortcomings, engineers assigned for the first time to permit the pilot to actually control the reentry vehicle rather than simply ride as a passenger as he had in the past.

The solution was found through a closed loop integration of the coverage with Johnson's Telemetry and Logos computer made by the Radio Corp. of America.

The coverage itself is powered by a 180-ton, 1,600 hp motor, has a 30 ft steel arm and gondola weighing a total of 84,000 lb.

In its overall result to overcome the problems of putting men safely into space, Navy is conducting:

- Spatial orientation programs at Naval School of Aviation Medicine at Pensacola



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Navy's known coverage of Johnson's Telemetry and Logos range, the first step in actual pilot control, is believed to be the world's largest.

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cola 1 $\frac{1}{2}$ to 2 minutes, and define human reactions, performance and the flight control capabilities of men in space flight.

- Bio-energetic program and a human elements of the Navy Medical Research Institute, Bethesda, Md., to determine human tolerance to the temperatures that will be encountered.
- Teletracking and recording equipment essential for transmission of physiological responses from space in order to make development at the Medical Research Institute, which also is studying the physiological and psychological stresses imposed in space flight.
- High pressure environment facilities at Bethesda and at the Naval Medical Research Laboratory in New London, Conn., are attempting to determine the role high-pressure altitude can have in impacting and packaging the habitability of space flight vehicles.

• **Flying Laboratory of Navy's Project RAMI—Research Aviation Medicine—**is studying and recording physiological information under field and in-flight conditions.

- Acceleration and space centers programs are under way at the Avionics Medical Acceleration Laboratory in Jacksonville to try and determine man's ability to withstand and perform under acceleration stresses and the human aspects of the reaction program.

• **Maintenance of artificial atmospheres and support equipment** is in the development of the Air Crew Egress Laboratory in Philadelphia. Navy shares the laboratory in the only one in the U.S. devoted primarily to this type work.

More specifically, here is a rundown of the space work now under way at Navy's four human factors centers.

Aviation Medicine School

Researchers at the Naval School of Aviation Medicine in Pensacola are simulating space flights should be designed primarily to ensure the absolute safety of the pilot with a sharp transition on communications control and data collection.

First, record flights, then see, should follow a ballistic trajectory. Following steps, officials add, should be to put a manned vehicle into orbit for a "new man time," then extended orbit and, finally, actual space operations.

The school trains pilot selection and training as a critical link of space flight and has established these basic parameters for a crew member—a minimum age of 18 to ensure mature judgment and emotional stability, a maximum weight of 150 lb. To gain further insight into the optimum training and selection of crew members, the school is now studying:

- **Could become orbital at 10,000 ft.**
- **Influence of socket matching on the pilot.**
- **Behavior of men in artificial atmospheres and in isolation.**
- **Influence of prolonged lack of gaseous stimuli on the locomotor center of the brain.**
- **Effects of prolonged weightlessness and artificial vestibular gravity.**

Research Institute

Work being conducted at the Navy Medical Research Institute includes experiments with high and low pressure chambers to study the relationship of space and biochemical laboratories to evaluate the chemical changes associated with physiological and psychological stresses that will be encountered in space.

The institute's physiology laboratories are studying the types and effects of human agents to be found in a conical atmosphere. Its physiological data tables laboratory has developed techniques suitable for space to earth transmission of physiological and psychological information.

Along with its pilot training now being used for N-15 research, the Avionics Medical Acceleration Laboratory has several special facilities. The centrifuge complex is used to study various complex acceleration patterns, and can

give half light simulation up to 40 G's. The laboratory would like to put into space at the earliest possible time a small animal laboratory to test the post-removal of a small animal undergoing an extended period of zero gravity. On the earth, zero gravity can be simulated for a few seconds at the most.

Other programs include testing and development of components capable of guiding, orienting and pilot restraint systems under acceleration.

The Air Crew Equipment Laboratory located at the Naval Air Materiel Center in Philadelphia is concentrating upon studies of extreme altitude environment habitation, space suits, cryo suits and associated equipment redesign, design breast and data presentation, in crew escape methods, general equipment and escape problems.

The laboratory is equipped with a vertical accelerometer and a horizontal accelerometer and has an explosive decompression chamber that can simulate altitudes of up to 100,000 ft. and temperatures from -50 to plus 150 $\frac{1}{2}$. It provides a maximum rate of 21,000 to 60,000 to 150 atmospheres.

Laboratory officials also hope to build a mockup of a capsule to provide the necessary breathing gases and logistics for space flight plus the removal of garbage, complex waste and vacuum systems for ventilation and radiation

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ND FACTS



Accuracy measured in millionths of an inch, made visible in the lower eye ball. The heart of New Departure precision ball bearings held to 2 millionths of an inch or less in accuracy. Graph at left shows capacity variation of a ball on the order of one millionth of an inch (100,000) measured by interfered machine. Graph at right shows eye ball. (2000X)

PORTRAIT of PRECISION!



The extreme accuracy of New Departure ball bearing component parts is now playing a vital role in successful vehicles for the Army, Navy and Air Force. Above — typical bearing parts; low capacitor — interfered photograph.

A mechanism is only as accurate and reliable as the bearings supporting its moving parts. For this designer the problem is how to achieve the material rigidity or accuracy of location, yet be assured of extreme freedom of rotation.

A "tip-off" to the solution lies in the short story—superprecision steel balls, the heart of New Departure precision ball bearings. For, with balls held to 5 millionths of an inch or less out-of-roundness and other bearing parts finished with comparable care, such bearings can be mounted and prelubed to provide the unimpeded accuracy of location and ease of rotation required of the finest precision instruments.

The AChievor guidance system proved in tests of the Air Force's Thor ballistic missile demands tolerances often measured in millionths of an inch, as in the case with the New Departure ball bearings on which the AChievor's precision gyros turn.



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EXPERIENCE BUILDS UP A BALL



ATLAS thrust measuring system (left) indicates and records value of thrust pushing and measuring force. Device at right reads inside propellant weight.

Three Firms Build Measuring Unit

Three subcontractors were teamed to build the thrust weighing and measuring unit used for launching Convair's Altus intercontinental ballistic missile at Cape Canaveral, Fla.

Convair itself did the design work on the static mechanism. It then subcontracted the work to Spitz Corp., a company which was formed last year and bought out the manufacturing activities of Shaw & Eddy Shaw & Eddy now operates only in the construction business.

Spitz is, turn, subcontracted the weighing and thrust measuring unit to General Industries, Cleveland. Ohio General bought the tool cells for the Cape Canaveral unit from Brillion-Luna-Hawkins Corp.

Buckhorn-Luna-Hawkins has built the entire equipment for Convair's Seaquester missile test site in California, but it was not responsible for the Cape Canaveral equipment except the tool cells—Anastasia Wren, said in its Feb. 24 issue, p. 26.

The Gilmore weight and thrust measuring system is capable of measuring and recording value of missile weight, propellant weight, thrust pushing motor force and thrust measurement force.

Force is transferred into values represented by weight and thrust, push, moment, raw moment, rail current, "XX" raw side load and "YY" raw side load. The system compensates for the effect of side load with respect to wind and with respect to extracting moment. A patent is pending for the system.

Four of the Gilmore systems are installed at Cape Canaveral and three are installed at a static test site at Edwards AFB, Calif., under the test vehicle testing procedure.

Besides the Edwards and Cape Canaveral installations, Spitz Corp. has installed installation of thrust measuring equipment for the Titan intercontinental ballistic missile static test complex at Mayport-Durham.



Rocket Engine Tested on Navy Fury

Ground view of F4U Fury fighter shows conditions for Redstone AR-8 rocket engine installation over test page. Rocket uses JF-6 and hydrogen peroxide.

PUMP PRIMERS



by Arthur A. Nichols

High efficiency liquid coolant pumps for electronic equipment

For electronic equipment operating at high aitudes, air cooling of high speed components becomes impracticable because there are density limits ceiling fan efficiency. Liquid cooling systems are therefore frequently preferred for this type of service. One electronic equipment over the years in the production of high performance aircraft engine pumps has been of great value in developing electronic coolant systems that provide maximum weight and space savings with efficient heat transfer capabilities.



Fig. 5. Electronic coolant pump. We have designed and built thousands of pumps especially well suited to electronic equipment coolant service for both airborne and ground installation (Fig. 5).

The ability to tailor pump configurations to meet specific performance and savings design conditions provide important advantages in the electronic equipment field with a problem of weight and space conservation. While this is of definite concern as usually all accuracy and minute individuality, it is particularly important in aircraft programs because the new equipment must be made to fit existing space. The availability of hydraulic pumps having unusual design adaptability, plus inherent flexibility and compactness in application makes them particularly suitable.

The feature of custom pump design not only means maximum simplicity in piping but permits further simplification of system design by incorporation of relief valves and other system accessories in the pump itself. Hydraulic pumps can be equipped with shaft configurations to suit drive requirements in accordance with the method of coupling best suited to the application. We can also supply using long motor driven pumps with integral assemblies of pump and motor.

Technical data is available and your inquiry is invited.

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Air hoisting XM-64, Navaho cruise missile by its Cape Canaveral, Fla., test stand, at left. Rocket-type booster has two 131,000 lb. thrust chambers. At right, missile and booster head upward at smoke obscures launch pad.



Navaho missile and its big booster rose, peak of sharp climb angle, shortly before witnesses are due to separate. XM-64 was powered by two Wright Aeronautical Division rocket engines.



Photos Show Navaho Firing, Separation Sequence



Navaho booster, pushing missile higher, develops 271,000 lb. thrust, is said most powerful of its kind. Project has been canceled.



Long-range picture shows start of separation of rocket booster. Before, booster troubles, burning in collection.



Last Navaho missile at end of test program was destroyed when rocket engine experienced jammed valve. Cape Canaveral range safety officials destroyed weapon over Atlantic Ocean.



PROTOTYPE Saab J35 Draken has flown at Mach 1.4, can go supersonic in climb. Tail pipe is partly visible on this model.

All-Swedish J35 Aims at Mach 2 Speeds

By David A. Anderton

Landing. Sweden's first production Saab J35 Draken afterburner fighter has been delivered to the Royal Swedish Air Force.

In a view, first signposts of the aerospace conglomerate are expected to become operational, replacing the Saab J29 "Flying Beech" now constituting the majority of Sweden's combat air force.

Further development of the Draken with Mach 2 performance and air-to-air missiles should increase its operational capability in the early 1960s.

Prototype Drakens, powered by a single Swedish-built Rolls-Royce Avon 200 turbojet with Rolls-Royce afterburner, have been flying at Mach 1.4 in high speed tests. Production airplanes, using the same engine but with a

Swedish afterburner making full use of the Avon's potential, will have top speeds around Mach 1.8.

Swedish radar equipment will be available in production deliveries to give the single-seat airplane all-weather capability. First airplanes of the line will have night-fighting capability only.

All-Swedish Plans

With the exception of a few accessories and components such as its Goodrich wheels and Collins avionics, the Draken is an all-Swedish airplane, the product of first class engineering and technical talent and of close cooperation between the Saab Aircraft Co. and the Swedish Swedish Air Force and Air Board.

The exclusive Avianon Wrenk away on the double delta Draken is the first detailed engineering analysis of the

J35's design development, flight test progress and production.

The Draken (Dragon) is a relatively small airplane, dimensionally comparable to the Douglas F4D Skyraider series, and probably also comparable in performance. Block plans have engines on the basic 10,000-lb thrust company both are relatively light and both look about equivalent on drag.

Overall length of the Draken is 46.5 ft and the wingspan is 30.5 ft. Depending on armament, gross weight is between 10,000 and 20,000 lb. Area of the double delta wing is 538 sq ft, wing aspect ratio is 1.77. Leading edge sweep on the outer wing is 60 deg, and on the inner wing panels is 57 deg.

Powerplant is built in Sweden under license by Svenska Flygmotor. Designated RM 6 by the Swedish air force, the engine is a Rolls-Royce Avon 200



STRUCTURAL changes in first production model of J35 Draken includes major cockpit modification. Tail pipe has been extended and a HF antenna is installed on belly. "U" designation denotes retract air winging experimental flight test.

series design probably equivalent to the RA 24, two of which power the English Electric P.1B. Dry rating of the RA 24 engine is 11,500 lb thrust, a true test figure. With a Rolls-Royce afterburner, its other figure increases to about 14,500 lb. These engines were installed on the prototype airplane, production craft will have the same basic engine but with a full afterburner developed by engineers of the Air Board and Svenska Flygmotor. Thrust with afterburning will approximate 15,500 lb.

Complete airborne radar gear will be installed, including a nose scanner for search and a scope for presentation of search and other target data to the pilot. Two-control radar will also be fitted. Standard armament configuration will be a pair of air-to-air missiles or two pairs containing unguided air-to-air rockets. A pair of 30-mm cannons can now be mounted to far ahead of the break in the wing leading edge.

Configuration Close-Up

The fuselage is in two major sections, joined by a belted hinge connection. Forward section includes the forward wing roots, cockpit, duct inlets, nose gear, main integral fuel tanks and equipment. The rear section includes rear wing roots, main landing gear, gunpod, lagging fuel tanks, armament and equipment. This is the major structural heart of the airplane, and once assembled, the only remaining structural weaknesses are two small wing outer panels and the vertical surface. Trailing edges of outer wing is blunt, starting at the tip with only the double-delta thickness and converging progressively toward to a small thickness of about two and one-half inches at the root.

The control surfaces for pitch and roll are at the wing trailing edge, and are in three regions—two inboard, two outboard. All three surfaces are deflected simultaneously, with differential deflections being used to combine roll with pitch. There is no flap tests on the Saab J35 research plane, posed there was no need for the surface.

Four pitot speed probes are mounted on the fuselage just ahead of the plane of the control wings.

Two modern hydraulic pits operate each surface. They are fed by two separate hydraulic systems in that if one is inboard out in parallel the other will get the pilot home. None of the atmospheric large moment gets back to the stick or rudder pedals. Instead, rock forces are artificially generated and added back to the pilot by a return that senses Mach number and dynamic pressure (Q).

Control tips are not cheap, although their status of emergency is quite important. They are separated from the fuselage by



DOUBLE-DELTA configuration and tail cone forward wing is shown in this prototype. Inboard panels, front outer delta wing, main landing gear and gun stability equipment. Chined metal nose may channel the flow.



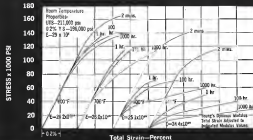
CLUSTER of J35 Draken fighters, as under each wing, can be carried on Draken. This version is 31.5 cm air-ground type. Inboard fence is at upper left.



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The AUTO ZOOM Lens for TV cameras using Vidicon tube tubes. Focuses from 1/2 inch to infinity. Zooming and focusing can be done remotely through standard drives.

Exclusive Series

Auton Work, with this article begins a two-part series on Sweden's first JSI Devel. represents, respectively, David A. Auton, Auton Work's European editor, and the first technical editor to see this article in its detail and in the first to make an out-of-the-ordinary report of what he saw. Next week, Part II will include an exclusive series of production photographs of the aircraft.

1. maximum depression of about two inches. The inlet duct system has an internal baffle that improves flow distribution at the engine face.

The JSI generally behaves like an out-of-the-ordinary fighter aircraft with high response performance. There are some detail differences that prototypes have been flown without stability augmentation, although a pitch damper on the autopilot circuit will be installed in production airplanes. So far there has been no need for a yaw damper, although this could be a later addition.

Center of gravity position is not defined in terms of per cent of chord for the JSI. Usual procedure is to keep it as far aft as possible; normal position is just behind the leading edge bend, in wing planform.

Normal travel of the CG during one flight is a maximum of four inches. For different load alternatives a total of about one foot is possible.

The only flight restriction on the airplane is in the case of the rolling platform, where the roll rate is still kept to a lower value than possible because structural limitations still are under study.

Stiffness behavior is a little different, after the roll at extremely high angle of attack, the JSI will lean at a high angle of attack in a stable attitude. If the angle is further increased the plane becomes unstable and then stable again.

Ground effect on a delta with such



FLIGHT test measurements are recorded by two Infrared microscopes which are installed in Devel's side nose section.

AVIATION WEEK, March 24, 1958

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ACCURATE, CONSISTENTLY RELIABLE AC output, proportional to linear acceleration, is provided by this new Giannini AC accelerometer. Available in ranges from ± 1 g to ± 32 g, the instrument has a built-in output of 4 volts which may be fed directly into a relay or other impedance with little or no phase shift.

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a low aspect ratio is very high and therefore the deceleration is high before and after touchdown. Like some other delta, the Daken is mentioned earlier, but on flaps.

Naval flight workers is to see differences from the start of takeoff until the Daken is well on and away. Operational pilots should be able to get off the ground in about 1,650 to 2,000 ft. at takeoff speeds between 150 and 175 mph. An experienced Daken pilot can reduce that run to 1,500 ft.

Climb and Maneuvering

Initial rate of climb is on the order of 30,000 to 40,000 fpm. Rate of roll is tremendous because of the airplane's small span, 100 to 225 deg. per second in the normal case used by Sab pilots.

With the three wing of the Daken there is little time change in the trans-

verse speed except. Instead that the plane will accelerate rapidly to its maximum Mach number.

Prototypes powered by the Avon 200 with a Rolls-Royce afterburner can hit Mach 1.4 in one hour, but the threefold lift coefficient in production airplanes will boost that figure to about Mach 1.5.

Approach speed is about 200 mph. The airplane is trimmed for a landing incidence of 12 to 14 deg. Touchdown is at about 177 mph, and the inverted landing roll without brake parachute with maximum braking is about 2,000 ft.

The landing roll can be greatly reduced. On a dry concrete runway, it is possible to land in a stop within 1,400 ft. of the touchdown point. On wet runways, it is expected to be able to land in less than 2,000 ft.

Operational airplanes will have a



Roll, distribution and flex is tested on the wheel. Daken is in vertical plane through thrust axis; 81/2" distance would be lost left to right in picture.



DRAKEN shows chamber takes case section and forward fuselage, run timidity aircraft and temperature extremes. Rubber shape shown above will be in production (31).

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body parachute to keep the landing run low.

Landing after abnormally high approach speeds proved one part of development flight testing. FVB brought the Dakota in over the fence at 191 mph, touched down at 165 mph, instead of the usual figure of less than 140 mph. On touchdown they popped the chute, slowed to full brakes and came to a grinding halt less than 2,000 ft from the fence. This is indicative of high performance under combat conditions.

Flight Testing

More than 700 flights have been relied up to date on the three prototype aircraft, now recently joined by the first production plane. Before then, the aerodynamicists at Sub 210 had made about 1,000 flights in the low-speed regime to check the expected flight characteristics of the double-delta wingform.

Flight of the first prototype Dakota was Oct. 25, 1955, and the second followed it into the air shortly after. First production airplane flew Feb. 15 this year.



MEASUREMENT centers are part and parcel of research here. Top and bottom, FVB can switch from one type of program to another during flight to achieve time to victory.



First prototype flew successfully without abnormally fast the first time on Jan. 29, 1956. Two months later during flight tests it went supersonic in a climb.

Normally the third prototype goes immediately to the Royal Swedish Air Force for evaluation tests, so that the company has been doing most of its flying with two prototypes. The third currently is undergoing the rigors of winter operations at Lulea, Sweden's farthest-north permanent air base just below the Arctic Circle.

Most of Sub's 16 test pilots under Regent Olsson—chief of flight test, have checked out on the J31. Six of these are rated as experienced test pilots, including Olsson.

Flight test data is recorded photographically and electronically with oscillographs and camera recorders. Teletyping has not yet been used, but the subject is a hot one in engineering discussion here.

Flight-test instrumentation is built, bought and built built at Sub for special purposes.

One example of aerodynamic flying brought out for early flight tests in the

Progress in Propulsion at Marquardt



Roy E. Marquardt, President

In all of the complex of modern weaponry no area affords more challenge than the field of supersonic and hypersonic propulsion. Here, where the stringent requirements for engine thrust, size and thrust are creating problems of critical consequence, Marquardt engineers and scientists are making continuous progress.

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Two Marquardt ramjet engines sustain the thrust of the Bomarc after it is launched vertically into the air. The ramjets provide a range that allows the weapon to destroy enemy bombers at a far greater distance than any other missile presently in use in air defense.

The Bomarc application is only one of the many propulsion projects now underway at Marquardt. Currently under development are several advanced supersonic ramjet engines for application as future weapon systems.

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from installation on the dety wings. These forms give more stability and maintainance and reduce belting considerably.

Carburetor leading edges aerobically look in the maintenance's bag, not under test on a prototype. Ductless Deter could have been built and should be ready for installation and flight about now, following the complete overhaul of the two component-operated prototypes during the build-up plan. But Sweden sells only rock you.

Such engines expect a 10-15% improvement in the lifting ratio with the carburetor wing. This aerodynamic improvement has shown up in the General F-102A with a single wing for. They also expect to eliminate the wing finches with the carburetor system.

Draken Design Details

The double-delta configuration of the Draken is its most unusual and recognizable feature. These were good aerodynamic reasons for the choice, and their strength from a series of theoretical calculations done by the design team under Erik Linde, chief designer for the J35.

Engineers originated during the two year period 1949-51 as a follow-on for Saab's J29 "Fighting Bear" intercepter. Basic design goal was to produce an intercepter capable of loitering down supersonic jet bombers. The Swedes, who live somewhat closer to Russian boundaries than the Americans, believed it probable that the Soviets there were operating some new light jet bombers. This was knowledge that took a longer time to be officially reported to the United States.

In 1952 the engine had been selected—it was to be a Rolls-Royce 280 series built under license in Sweden by Svenska Flygmotor. The design team began the task of making more detailed layouts, setting equipment and pilot in place.

Four men studied the preliminary design of the Draken. Bert, who is an engineer and a pilot, one mathematician, and two aerodynamicists—one a supersonic specialist and the other a subsonic specialist.

Flight envelope for the proposed intercepter cut through Mach 1 at low altitude, and continued to increase in speed to a maximum at 50,000 ft. Above this level the speed dropped off to where the limitations of engine, fuel and airplane set the maximum value. From there down to sea level it followed the usual curved line between altitude and sea level stall.

Wing ones and spurs are affected by airfoil requirements as well as by maneuver and recovery. A series of aerodynamic calculations taking those



North American's UTX Jet Utility Trainer—the "Dukehound"

The UTX comes out

The fuselage of North American's UTX is out of its jig! In a few weeks the plane will be completed and early this summer it will thunder into the air.

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Wind tunnel tests for the new craft were conducted at the CWT—the Southern California Cooperative Wind Tunnel. Since 1945, the CWT has been responsible for the aerodynamic testing of a host of military and civilian aircraft—among them the X-15, the DC-8, the Starfighter, the Mustang.

The CWT serves its five owner companies, a number of governmental agencies, and other aircraft manufacturers. If you would like more information regarding its services and facilities—or employment opportunities—you are invited to write us.

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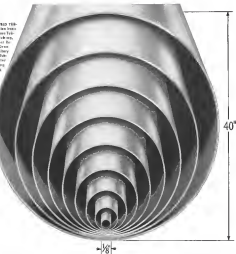
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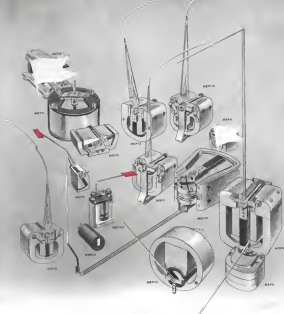
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Liquid Oxygen Tank for Thor

This liquid oxygen storage tank, part of the Thor ground support complex, is shown on its way from the manufacturing plant of Canadair Corp., in Lowell, Mass., to Henshaw AFB, Bedford, Mass. From there, the tank was to be flown to Cape AFB, TX. Designed to hold 15,000 gal. of liquid oxygen at -297.4°F , the tank is 50 ft. long and 5 ft. high. Built along the lines of a Thoros bottle, it consists of an inner tank of stainless steel and an outer tank of aluminum. The space between the tanks is a vacuum and serves to insulate the liquid oxygen. The tank weighs 52,000 lb. empty and over 130,000 lb. full. Its own wheels give the unit mobile stability. Canadair Corp. is a wholly-owned subsidiary of Canair Corp., holds similar equipment for Atlas, Jupiter and Redstone missiles.

factor into account gave a wing area that was too big and had to be cut down. The Sub contractors didn't want to reduce the span because of the reduction in control that would follow. They had already selected a delta as a result of experience and lateral considerations, so the only thing to do was to chop area out of the delta geometry. This could best be done at the leading edge.

But such cutting brought a disadvantage, the unusual position of the center of pressure on a delta wing is fairly far forward of where a designer wants it.

The center of gravity of a delta is too far aft of where he wants it, so that the designer has to give its length forward for balance.

Chopping into the leading edge moves the center of pressure aft. The CP CG relationship becomes conventional. Compared the J15 with other delta-winged aircraft in service, one shows one result of the double delta.

The Dakota did not get remembered at birth because Watson's theory had not yet been developed. Later on after the theory was fixed, Sub contractors got the word and immediately checked the layout of the Dakota like some delta-winged aircraft. The size distribution wasn't bad, because the wing thickness is well distributed and the maximum thickness is well at the Dakota's point with a reasonable

airfoil shape and didn't require tailfins.

The size distribution is more favorable when the J15 is carrying weapons externally, such as the pair of air-to-air missiles in rocket pods that are to be part of its standard intercepter armament.

Project Growth

At the time of J15 preliminary design, there was less exposure. By May, 1959, the group had increased to seven designers and three structural analysts.

The engineers knew there would be an advantage in having some full-scale flight tests of their external layout, particularly at the low-speed end of the flight spectrum where handling characteristics assumed tremendous importance. So they designed the Sub 213, a flexible little bellows model of the Dakota geometry. The design and drawings took nine months and the shop crunched that figure in construction. The Sub 218 was on Jan. 21, 1962, and made about 1,000 test flights that turned in some of useful data for the designers. Now, already in progress, it is in use in front of the flight test building.

Later in 1952 the detailed design had started further growth of the team to about 10, and in succeeding years the design group has grown to the current total of about 150 engineers, not

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including looking, looting, stoves, and weapons group. This is a remarkably low number of people to have landed and defend against a high-sophisticated airplane.

Prototype design started in May 1952, and the first drawings were released to the shop in October, 1953. Two years later the first prototype flew.

(This is the first of two articles on the Sukh 115. The second will appear next week.)

Swiss Vote for P.16 As 2nd-Stage Fighter

Geneva—Swiss Parliament Lower House has voted to order 180 Swiss-built P.16 ground support planes as the second stage in the replacement program of the Swiss Air Force.

First stage was the purchase last November of 180 Hawker Hunter at a cost of about \$74 million including spares and support (AW Nov. 23, p. 27).

The decision on the P.16 order was reached after a heated three-day debate which reached its climax when a vote of 111 against 36 in favor of "buying Swiss" was taken.

A second vote of 131 against 7 demonstrated that the response 180 Armstrong Siddeley Sapphire engines, plus 55 spares, should be built in Swiss lead under license.

Cost is estimated at \$101 million of which \$7.95 million will be charged to the buy-sell-out expense.

The whole matter is now subject to Upper House approval.

The Flugzeugfabrikwerke A.G. (FFW), Altwies, got the production order with delivery scheduled to be completed by the end of 1962, but the government-sponsored federal aircraft works at Emmen (Edg Flugzeugwerk) will produce production including final assembly and flight test. The P.16 was initially attacked throughout the debate for never having kept its promised delivery date.

Agreement against the high to cost German Super Tiger (T119 SP) was:

- It is too expensive. In January, this year Germany was asking \$130 million for 100 of this type. Now to produce these under license would still cost about \$140 million total.
- Only prototypes of the modified version suitable for Swiss requirements are available. It is primarily a fighter and to adapt it to ground support work would take at least two years of further development.
- Its takeoff and landing performance is not short enough for Swiss conditions.

Toward the end of the debate, the words of the German Super Tiger

were again emphasized and it was recommended to continue present negotiations with Germany with a view to equipping the Swiss Air Force with a series of Super Tigers as the next stage in its replacement program.

French Continue Cuts In Orders for Aircraft

Paris—As a result of a 25% cut in the 1958 French Air Force budget, several orders for new aircraft were reduced sharply, while production rates on other aircraft orders were slowed down.

Aviation Week reported previously

"AV March 1, p. 27") that various prototypes such as the Latham C-22 was cut and Sud Aviation's supersonic bomber, Super Vitesse, has a better chance of being cancelled. Other cuts included in National Assembly budget debates in March.

• Dassault Super Mystere. Original order for 170 aircraft was reduced to 120 and the production rate was reduced from 15 monthly to nine. Super Mystere production models are just now coming off line.

• Breguet's original order for 160 of its subsonic variable technology aircraft, Alouette, was reduced to 75. Last production Alouette aircraft is just coming off

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line. Request also led to air force order for 15 low engine Sabana cargo aircraft cut to four.

• Navy will delay placing an order for 180 Dashless E-100s.

• Fagga Magister jet trainer production order will be cut back to six months from 12.

• Sud Aviation Vautour production rate has been reduced from nine to four. Original order for 160 Vautours, to be delivered in three versions—bomber, all-weather fighter and attack—was cut back to 100. Most will be all-weather and attack versions.

French will sell about 25 of the 100 to the Israeli Air Force.

• Sud Aviation has had an original order for 320 Dague helicopters cut back to 120. The original order for 300 Alouette was sliced to 150.

In addition, 150 Sikorski S-58 heavy helicopters, which were to be built by Sud Aviation under license, were cut back to 80. Production rates on both the Dague and Alouette were cut just about 50%.

British Will Spend More for Missiles

London—British Government will step up its spending on missiles during the 1978-79 fiscal year while cutting back purchases of conventional aircraft.

The government reported to the House of Commons last week that overall spending under the 1978-79 estimates will be \$259 million less than in the current fiscal year.

"We are spending less on aircraft," a spokesman said, "although we shall be backing up the deterrent with Victor and Vulcan and introducing the Mark 2 jeton for its defense." He said the decrease in expenditure on aircraft is partly offset by increased spending on missiles, armaments, radar and weapons of all kinds. Spending on satellites and space is down 21% while on the other categories it is up 50%.

First production deliveries of the

T-37 Engine Award

Award of an \$1,152,272 contract to the Continental Aviation & Engineering Co., Detroit, for jet aircraft engines has been awarded by the Air Materiel Command, Wright-Patterson AFB, Dayton, Ohio.

The engines are for the twinjet T-37 Air Force training plane.

The Detroit contract was one of three totaling more than 321 million awarded. The other contracts went to United Aircraft Corp., Stratford, Conn., \$26.1 million, and Ryan Aeronautical Co., San Diego, \$1,152,272.

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be Hamilton Presses are scheduled during the year and service trials of the Bristol Bloodhound surface-to-air missile will begin within the next few weeks at North Coates.

Twenty-eight million dollars will be spent on American missiles during the year, it was reported, an expenditure documented by expenditure spokesmen.

Total spending under the 1955-59 air estimates is to be \$1.8 billion, \$50 million under the current year's estimate. Because there is no U.S. aid in the new estimates, and because German contributions to maintenance of the Royal Air Force in Germany will be less, overall reductions amount to the \$199 million figure.

Other points made during the air debate:

- English Electric's P1 will come into service with lighter armament during the year.

- Jeton Mark 7 and Mark 8 are to be equipped with afterburners to extend their effective life.

- Blackburn's NA 59 experimental naval strike aircraft, powered by two de Havilland Gyron Junior turbojets, is being considered for the Royal Air Force as well as designs put forward by industry in reply to a general operational requirement for an aircraft to replace the Canberra.

- Discussions are underway with the Army on a successor to the Blackburn Beverley transport.

- Some 268 V-bombers have been ordered, including 77 Valiants, 75 Vulcans and 195 Victors, of which 160 aircraft have been produced.

- Bolkow's V-bombers are capable of making the roundtrip to Moscow with out aerial refueling.

- Work is almost complete on V-bomber bases and the water clock.



USAF Housing

Model shows USAF officers quarters to be built at Moffett AFB, Calif. Designed to combat rising land costs, housing would be eight stories high and would house 480 men. Construction is said to be of exposed concrete with aluminum windows and doors. Cost is put at \$8 million. Priority is designed to the urban Congressional legislation of \$7.800 per man. Unit is prototype for future construction at other air bases.



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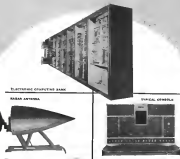
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USAF Seeks Weapon Environment Data

By James A. Faxon

As Force weapon systems of the future will have to live and fight in all theaters varying from the present operational ceiling of about 75,000 ft. to outerplanetary space itself.

Natural and induced environmental conditions encountered at these altitudes, called lower measurements, are being studied for USAF by Radio Corp. of America to assess suitability of future weapon systems through extent of available knowledge and its development of scientific techniques for integrated testing of subsystems, equipment and components.

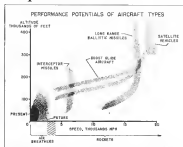
Philosophy of the study has been to develop ideas for design of terrestrial functions capable of operating over the limits of the different environments wherever possible and which will be applicable to many types of equipment and systems.

Study is being conducted for the Environmental Criteria Branch of USAF's Wright Air Development Center by RCA's Defense Electronic Products Division. First part of the study—suitability and analysis of available data—has been published in WADC Technical Report 57-496, by R. A. DiTanto and J. J. Luech.

Future Weapon Systems

The purposes of the study, future USAF weapon systems are classified as:

- **Sustained flight vehicles.** These are the powered vehicles deriving most of their lift from aerodynamic means (the remainder from centrifugal force). Minimum altitude and flight is airspeed defined as 150,000 ft.
- **Boost glide vehicle.** These are vehicles which generally are powered for the first few seconds of flight, rise above the atmosphere, and coast through glide, and land continuously (using lifting surfaces). Altitude and speed limits for this vehicle are considered to be from 150,000 ft. and 4,000 mph. to 100,000 ft. and 15,000 mph.
- **Ballistic vehicle.** This vehicle is powered during its first few seconds of flight, runs above the atmosphere, and follows a ballistic trajectory to its target, not contributing aerodynamic lift. Altitude and speed limits for this vehicle are considered to be 75,000 ft. to 11,000 mph and 45,000 ft. to 15,000 mph.
- **Satellite vehicle.** These are vehicles placed in an orbit so that their velocity



Predicted operating ranges of future sustained and sustained boost and glide vehicle, according to National Advisory Committee for Aeronautics, is gradually gain and during first few seconds of flight, after re-entry, it approaches target in hypersonic glide.

generates a centrifugal force equal to the earth's attraction. Altitude and speed limits of this vehicle are 150,000 ft. at 20,000 mph and no upper limit. These future weapon systems will be exposed to two kinds of environmental conditions: those present naturally above the present altitude, and those created by the existence of the vehicle and interaction between the vehicle and the natural environment.

Environmental conditions generated by operation of the vehicle might include shock, vibration, and acoustic conditions due to air intake, landing or propellant operation as well as operation of rotating or oscillating equipment within the vehicle.

Interaction with the natural environment could cause such effects as vibration, acceleration due to aerodynamic effects, high acoustic noise, high aerodynamic effects, and high temperatures resulting from high speeds.

Specific induced environmental conditions are:

- Temperature
- Accelerations
- Vibration
- Shock
- Zero gravity.

Induced environmental are highly dependent upon design of the particular vehicle in question. Induced environmental conditions, for example, is dependent to a large degree upon weight of the vehicle, cooling system shape, size, shape, altitude, velocity and boundary layer conditions.

Natural Environment

Although much moderate acoustic data is available in the natural environment, conditions encountered at high altitudes and more is being gathered by use of the Unmanned Grouping Post and mobile projects. Actual knowledge of these environmental will best be gained from a relatively long duration, will be atmospheric vehicle.

Natural environmental conditions in the study are:

- Atmospheric composition.
 - Extreme high pressure.
 - Solar radiation.
 - Ozone.
 - Deionized gases.
 - Aerosols.
 - Ionized gases.
 - Solid particles.
 - Magnetic field.
- Two types of temperature exposure were also found in aircraft, ground and



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GROUND-SUPPORT CABLES and plugs, because of the self-aligning feature of the new Cole Connector, can be made as a quick-action type, hinged and toggle-locked, replacing time-consuming screw-on types that create threaded problems.

satellite flight. Thermodynamic heating occurs as missiles in satellite rockets pass through the atmosphere into space and during all high speed phases of a guided missile flight. Satellites in orbit, on the other hand, will be subjected to radiant heating from the sun.

Present state of the art of aerodynamic heating permits calculation of the temperature of boundary layers of an air at high velocities. At altitudes to 750,000 ft. in Mach 4 flight, temperature of boundary layer air is about 2,400-2,600°F. At Mach 6, temperature is 3,100-3,300°F.

These temperatures, however, are not the skin temperatures of the vehicle, skin temperature is a function of cooling, heat sink, radiation losses and time to the high temperature measurement.

In the sustained flight vehicle with a mission time of one to two hours, predicted maximum skin temperature is 2,000°F. Internal temperatures will vary as a function of design but the proposed value for external ambient temperature is 950°F with an initial rise from -45 to 950°F within 10 sec.

Boost glide vehicles although capable of generating temperatures beyond the melting point of structural materials, have been assigned a nominal value for external heating of 2,000°F. Both external and internal temperatures will vary with the flight path of the vehicle, although the vehicle remains very below 150,000 ft. temperatures should remain about constant as they do in sustained flight.

Ballistic vehicle temperatures should approximate those of the boost glide vehicle except that the flight path initiates heating from the ground up about 100,000 ft., maintains cooling through apogee to reentry. Temperatures at reentry should be kept below 2,000°F eventually and 950°F internally.

Satellite would be protected from high temperatures on ascent by an artificial nose cone. In orbit, normally uncontrolled temperatures would be -40 to 175°F.

Acceleration Values

For all types of vehicles, values of acceleration to which they are exposed probably will be greater to ground launch and on reentry and landing. Above 75,000 ft. and below 100,000 ft., intercepted accelerations are less than 10G's. Re-entry accelerations of 5G's for one second are anticipated for the boost glide vehicle. The ballistic vehicle must attain 60G's deceleration. Now high G values are expected above 100,000 ft.

Some of the dynamic effects resulting from accelerations are higher than actual effective pressure are induced on fuel and hydraulic lines, etc., when with their contents oriented along the direction of acceleration are not pulled out by the strap, acceleration

may cause vibration or change characteristics of detonating devices.

All vehicles will be subjected to static and dynamic variations in the static and dynamic pressure of which will amount to be determined by the power plant used.

Higher magnitude of vibrations occur during boost or launch, lasting about 10 sec. Aerodynamic variation due to the propellant will be very great on the ground and at launch because of the higher power requirements of these vehicles. This effect will be assessed by ground selection.

Aerobic variation also will increase with speed. At the vehicle reaches alti-

tudes where air density is very low, the aerodynamic aerodynamic forces decrease and the propellant accounts largely is not supported by the air in the rest of the vehicle. At these altitudes the propellant mechanical expansion still exists, and all surfaces which are being exerted increase their amplitude of vibration because of the lack of air damping which takes place in denser air. Finally, the skin, due to aerodynamic variation is caused by the vibrating surface and is dependent upon the density of the air within the skin structure.

In general, shocks encountered in transportation and handling of missiles and their equipment are minor pains.

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AMERICAN WEEK, March 34, 1955

INTELLIGENCE



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thus an actual operation. Highest magnitude of shocks has been the result of irregular burning of solid propellant, broader separation and failure circumstances.

Above 75,000 ft., operational shocks would be due to booster separation, air bursts, by canopy action and wind gusts. Booster separation shocks as large as 100Gs are possible. Wind gusts will cause shocks dependent on gust intensity and on the speed and rate of the vehicle. Above 100,000 ft. air-ion outgassing shocks are anticipated.

Zero Gravity

Future vehicles will be traveling at speeds where part of the lift comes from the force of gravity will be due to centrifugal force. As with orbiting satellites, outside equipment may have to function in a zero-g environment.

Of all the high altitude effects, probably the surest gravity conditions exist

near to some of the most essential efforts. These include:

- Air bubbles may not rise in batteries and so may contaminate the plates, reducing or stopping the flow of current from the battery.
- Electrical sensitive devices may not operate correctly.
- Pumps designed to operate on the ground may not work because hydrostatic heads present on earth would not exist in a zero gravity condition.
- No headway for some measurements of air. Heat transfer due to convection will not take place as hot gases won't rise.
- Cleanliness in space oriented equipment will change.
- Devices working on fluid level will not operate properly.

(This is the first of a series of two articles on high environments. The second, describing results of analysis of natural environments, will appear in next week's issue.)

Cornell Klystron Lifts Microwave Pulses to Record Power Radiation

Bottle-Microwave pulses with a peak power of 13 million watts, believed to be the highest peak power ever radiated, has been demonstrated by Cornell Aeronautical Laboratory by means of a specially adapted, continuously pumped klystron.

Performed under a research contract with Army Ordnance, the achievement is significant to the future development of ICBM detection equipment. Army says Project, however, was without funds for three months during the fall before a budget-cutting period.

Klystron developing the large peak power has been adapted from klystrons used by the Stanford University linear accelerator and built by Varian Inc. (the requires continuous pumping to maintain sufficiently high vacuum).

Problems of detecting an ICBM at sea level with a relatively low radar cross section at very large distances requires use of both of the following important to conventional radar.

- High power. Detectability of a war head is a direct function of the average power illuminating the target. Average power or target can be increased by increasing the length of the radar pulse as is done with Lincoln Laboratory's Millstone Hill radar, or by increasing the peak power transmitted as would be done with a radar utilizing the Cornell technique.

• Signal analysis. More than a dozen companies throughout the country are working on techniques for "fine structure analysis" of radar return signals. These techniques include Coker at the University of Connecticut (U.S. Air Force JORDIR) (AW Aug 19 p. 25)—

to amplify the received signal over a sufficient period of time to make it strong enough to be below the noise level of the receiver to a level where it can be analyzed on a cycle-to-cycle basis.

System developed by Cornell consists of the large Varian 5-stand klystron, vacuum pumping equipment, high power modulator, control console, plus a waveguide run to an eight foot parabolic antenna on the roof. Transmitted pulses are two microseconds in length, with a fill cycle rate that brings the average power to about three kilowatts.

Primary problems faced in the project was coupling the generated microwave energy through the waveguide and an antenna system into free space. To prevent breakdown of the 5-stand waveguide at the high power levels under study, waveguide is pumped down to a suitable vacuum.

Special care had to be developed. One of an initial approach exploited consisted of using a spherical-shaped "iris" to couple the waveguide feedhorn to air, where the feed parabolic antenna beams the energy into space at about a 45 deg angle.

Scientists from the Cornell Aeronautical Laboratory and that many experts in the microwave field held the opinion that the highest usable power that could be generated was about 10 megawatts, and that therefore the Cornell project was impossible.

Most logical construction of the project would appear to be raising of the average transmitted power and reducing problems involved in applying the technique to an operating radar.

APRIL 21st AIR TRANSPORT Facts and Figures

(PUBLISHED BY THE AIR TRANSPORT ASSOCIATION)

AVIATION WEEK has again been officially designated to publish "Air Transport Facts and Figures," compiled by the Air Transport Association. This editorial feature will appear in our April 21st issue and will provide an impressive picture of the tremendous progress achieved by the carriers during the past year. We are particularly grateful that this vitally important editorial feature is again available to AVIATION WEEK's world-wide audience.

AVIATION WEEK average net paid ABC circulation June-December, 1957, 67,008. Paid circulation of current issues: 70,178.

Current print order 73,039 copies.

For advertising message scheduled for the April 21, 1958 issue will be based to gain maximum attention from all those interested in the Air Transport industry. Regular advertising rates apply.

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CESSNA 175, company's newest addition to its business line. Four-place airplane, powered by Continental CO-300-A ground engine, is perfectly suitable to high altitude flight. New timing gear improves modification engine uses.

Aviation Week Pilot Report:

Gearing Sparks Cessna 175 Performance

By Robert T. Strafield

New York—Power and efficiency gained by its overhaul, 175 has Continental CO-300-A ground engine used. Cessna's new and recent addition to its business line, the single engine, all metal, four place Model 175. Power rated engine adds 75 to 1 psi per rpm generates 5280 rpm at 2400.

Reduction gear for propeller does not run at three fourths of engine crankshaft speed (propeller gear speed of 3,400 rpm at rated engine speed of 5,280 rpm). Maximum rpm can be increased to 10,000 ft. Fuel grade is 80/87 octane. Dry weight of engine

is 314 lb. Compression ratio is 7.1:1 and max. rev. displacement is 301.57 cu. in. characteristics ordered during flight overhaul by Aviation Week included:

- **High altitude performance.** At 10,000 ft., pulling 20% power, 3,150 rpm—engine with full complement indicated 150 mph for TAS at 150 mph.
- **Rate of climb.** Efficiency of McCulloch development propeller, 54 in. dia. with ground clearance of 82 in., varied during takeoff and climb at maximum gross weight 1,330 lbs. at take off 1,100 lbs. at 4,000 ft., 780 lbs. at 7,000 ft.
- **Noise level.** Engine sound is noticeable.

able modified. Cessna has reduction to ground engine principle, plus its new timing gear suspension in which cord is attached to chordal aluminum ring located in four engine mounting points, instead of being rigidly fastened to airframe. Cord fasts freely over edge of firewall.

Market Range

With base price of \$10,995. Full Wichita, 175 is rated at market listing between Cessna 175 and models 180, 182 and the Skylane.

Aviation Week. With all the production line, was N91548. With its instrument dual controls, good for long (about 1000) full fuel load and



ENGINE-TYPE engine modification adds to 175's styling. Control is fixed by its position with its additional seating by hand. New part enables flight maintenance on left side, controls center, engine maintenance right, takes on credit line.





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line of no aboard 175 ground out of maximum—2,350 lb.

New joint scheme enhanced 175-shk. line. Balde top, contact read should added to it—wing. To not slow down and seal out elements. Coors designed top of instrument panel with built-in receptacle to receive battery of windshield.

Windward inspection also showed thoroughness of company's and inspection. Feature can be extended to all of Coors line. American Wire has told. Engine mounting points holding used are connected to rubber shock mounts. These are braced directly to the engine mounts. Built-in shock engine mounts to firewall supports rubber shock mounts is aimed at reducing engine noise and vibration. Both the upper and lower end may be covered with periscope.

Oil cooler, standard equipment is essential directly on engine. Carburetor is Marvel NA-1 fast type. Intake system is perfect through of pump. Static test results are excellent. Provision is made for optional installation of engine-driven vacuum pump. But, it is not of luggage compartment and is accessible through panel door.

Weather was favorable for inspection. Wind was from the NW, at 15 kt. Sea level pressure was 30.16 in. (30.17 in.). Outside air temperature was 74°.

Engine was bonded to cut step up to 100 in. From rubber siding was one fuselage—one had two big ways in the back—and seats could be adjusted back and all the ease of flying.

Flight Check

Cockpit check showed fuel system valves on floor between front seats. Interposition valve—left, right, both and off—was positioned in 'both' normal position for takeoff and landing. Starting procedure as simple. With electric engine, engine fired quickly without any prime. Engine instruments located on right side of new instrument panel next right to the panel.

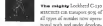
New panel mounts all flight instruments on left side, directly in front of pilot. Ranges are mounted on center line of right side paneling. Electrical switches are grouped top center. Lateral controls, including throttle, brakes, clutch, fixed gear, an electric, foot pedals and switches are easily read and accessible to both pilots.

Taxing was smooth and controls very good. Light pressure was required for turns, even over rough terrain. Nose wheel has of handle 30 deg. left or right and is hooked into rudder control.

Before taking turns, wings were elevated at 1600 rpm and we were ready to go. We had trimmed was down a few degrees left one flap up

Feet of Hercules, No. 2

New missile muscle now for NATO bases!



more for the U.S. Armed Forces. The maximum velocity while carrying some 1000 lbs. of cargo is 120 knots.

The rugged "go anywhere, load anything" capabilities of America's first group of combat cargo aircraft add new strength to NATO supply lines—or a time when these fire-fighting birds are more vital to the free world than ever before.

In addition to "leading off" the global mobility of the Armed Forces, the Hercules is making new friends for the United States by its prodigious loads. Example: a C-47 of the 32nd Air Division, 17th Troop Carrier Wing, Tropicana-Favelle Air Base, France, transported a 25,000-pound generator from Lyon, France, to where it was badly needed to supply electric power in Adana, Turkey—and flew the 191st tactical cargo company in less than seven hours.

Like all Lockheed designed and built aircraft, the C-47 Hercules can be counted upon to do its assigned jobs with maximum ability, to make how difficult the task. That's why—

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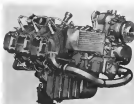


60,000 hours of service and over 4,000,000 trouble-free miles to 22 Bell Helicopters using 32 engines! 360 C. & L. Spinks, Superintendent of Maintenance for Petroleum Helicopters, Inc., has this to say: "AC 14-6's are the best plugs available for our Bell-Boeing Bell equipment!" From initial concept to improved profits, Petroleum Helicopters' Bell using the petroleum helicopter operators make the most service demands on spark plugs — steady speed and high temperatures over prolonged periods. And AC Spark Plugs meet these challenges. They keep away fouling carbon and oil deposits so long as they burn — stay cleaner longer! AC Aero-Spark Plugs deliver the high conductivity spark so necessary for peak power and efficiency. For better performance and longer maintenance, install AC Aero-Spark Plugs in your plane, TODAY!

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CONTINENTAL GO-180-A, six-cylinder, 175 hp, ground engine is rated at 3,100 rpm. Cessna's is geared to propeller shaft. Rotax's goes into group H (four-cylinder engine models) and speed. Rotax (right) is an all-in-one engine.



—Cessna recommends 26 deg. of flap with zero wind—and applied full power. Airplane went smoothly down run way, then reduced nose level and ran out.

Nose wheel lifted at about 40 mph with little back pressure. We took airborne in slightly over 300 ft, initial climb was at rate of 1,500 fpm.

Despite the load there was no strain on climb. With 36 mph in the suggested climb speed, we found straight-ahead visibility too limited because of a swirling eddy.

We varied climb between 30-100 mph and got best results: good rate of climb and visibility.

At 4,800 ft, we had 1,100 fpm rate of climb at 90 mph. Reducing this speed and maintaining 2,500 rpm, we held 700 fpm, then 5,100 and 7,000 ft. Leaving 9,000 ft we dropped back to 53 mph indicated and held 100 fpm until an level of 10,000. Through-out climb Cessna 175 was easily stable; only slight trim was required for hands-off flight.

Hands-Off Trim

At 10,000 ft, outside air temperature was 11F. With outside heated back, airplane trimmed hands-off and with power at 700-3,100 rpm—our TAS was 150 mph. Airspeed in indicated at 176 mph. Cessna's low roll-over rate from 140 to 176 mph. Performance at this altitude was excellent. Here we checked what Cessna representatives termed "zero trim stability."

Putting 175 into moderate turn, and adjusting trim, the controls were neutralized. Airplane continued hands-off rate of turn, holding speed and altitude. Through 360 deg. over there was no variation or banking. Retarding power to 2,500 rpm, level flight, flap down, speed was reduced by 48 mph indicated (45 mph TAS). Stability in

creased good at this speed and there was little stress in holding airplane level.

We ran through a series of power on, power off stalls, with and without flaps, at 10,000 ft.

With power off, no flap, the airplane was pulled up straight ahead, followed by climbing turn to left and right. In

each instance still warning rounded at about 55 mph indicated and light banking preceded stalls, which broke at about 45-47 mph. There was no clean break and sudden loss of altitude, rather a case of banking and break away, with altitude loss of about 150 ft.

Web had flap airplane stalled between 30-45 mph IAS, again preceded

Cessna 175

SPECIFICATIONS AND PERFORMANCE

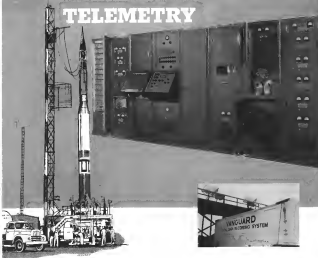
	Continental GO-180-A (6 cyl) hp	170	175
Propeller	McCulloch Fixed Pitch Model	PC PA-15	PC 84-50
Speed	Maximum (1) Sea level mph	107*	109*
	Maximum Recommended Cruise (2) 10,000 ft, mph	100	110
	Maximum 75% power, 10,000 ft, mph	100	110
Range	Range (3) Maximum Recommended Cruise		
	Wings based on 43 gal. useful	510	560
	Wings based on 43 gal. useful	4 3	4 7
	Time to climb (4) mph	120	110
	Maximum Range (10,000 ft, no reserve)		
	Wings based on 43 gal. useful	720	800
	Wings based on 43 gal. useful	7 3	7 8
	Time to climb (4) mph	100	110
	Rate of climb (sea level), ft. min.	800	800
	Service Ceiling, ft.	12,000	12,000
	Takeoff ground run (sea level, zero wind, 20 deg. flap)	710	
	Takeoff ground run (sea level, zero wind, 20 deg. flap)	1,340*	
	Landing ground roll (sea level, 40 deg. or full flap)	200	
	Landing roll (sea level, 40 deg. or full flap)	1,450*	
	Empty Weight, lb.	2,100	2,100
	Empty Weight, lb.	2,100	2,100
	Luggage Capacity, lb.	100	100
	Pass Capacity, per	20	20
	Pass, ft.	36	36
	Wing Area, sq. ft.	28	27
	Wing Loading, lb./sq. ft.	170	170
	Power Loading, lb./hp	33 4	33 4

* 175 lbs.

1,500 rpm

** Limited by max. allowable rpm

TELEMETRY



The Vital Link in Missile Progress

Missiles are test fired for only one purpose to obtain data that will help build better missiles. If the test does not yield this information it must be considered unsuccessful—regardless of how well the "test" performed.

Reliable telemetry equipment consequently occupies a vital role in the development of the missile as necessary to our defense program. There is no other way to collect and preserve the all-important data from unmanned and unrecoverable test vehicles.

Radiation, Inc. is a pioneer in the design and devel-

opment of advanced telemetry and data processing systems. A significant example of this work is the Vanguard ARB (Astronaut Recording and Reduction Facility) installation. This equipment provides Navy scientists with final reduced data on the performance of a Vanguard vehicle in less than 72 hours after a firing.

From this elaborate transmitter to complete ground stations, we have the experience and facilities to solve your problems in telemetry and associated areas. Write today for our brochure describing this capability.



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replace and checked and tested off when 150 ft.

During operations work—varying from an hour to 35,000 ft—fuel consumption was about 7.5 gal/hr.

Airplane no. five was equipped with Group 28 instrumentation: Navo Supersensor (N111-3 with arm, attitude, trim and bank, etc. of climb, altitude, altitude, clock, outside air temperature gauge, air speed and landing light (all lights installed in left wing). Cost: installed at \$1,100. Weight: 25.5 lb.

Dead controls, an optional item, are good at 507.70 liquid damage, which Group 28 controls add another 1 mph to speed, run \$195. Total cost of N111B approximately \$12,537.

Group 28 offers eight other instrumentation packages in addition to Group 28. Technical ratings these are:

• Group 1, including basic instrument of 28. However, in place of Navo Supersensor is Low 137A-V145 with Omnisensor (open crystals—121.1 through 127.0), and Navo Supersensor VC 27 (25 crystals—118.1 through 126.7). Cost \$2,150. Weight 16.5 lb.

• Group 2, which includes some parts but in place of 28's Supersensor is Navo Omnisensor Mark II (24 crystals—115.0 through 121.0 low band). Cost \$1,615. Weight 32 lb.

• Group 3, which consists each of basic instrumentation of 28, status Supersensor, along with other special and low frequency, antenna and mast.

Small variations are offered in other packages such as Group 2L, instrument frame which is similar to Group 2, except that Sensor 3547R, replaces Navo Omnisensor. Cost \$1,198. Weight 44.5 lb.

All radio instrumentation includes color speakers, microphones, headphones and various antennas.

175 on Boats

Group 28's up to instrumentation Model 175 with Boat fittings for April delivery. Tests have been completed at Grand Lake, GRU, and firm is waiting CAA approval for float operation.

Supersensor will test on new Edo floats model 115 2500, under in design to the 240 2570 floats used on the 150 Edo float has seven watertight compartments. Each alone will support 2,200 lb. float water displacement at 175 ft. gross weight of 2,540 lb. The float is scientific instrument, of about 90% in ball-balanced airplane.

Edo representative told Aviation Week that float would be floated in trials with two costs of one ship made pump. Eventually there would be price cost of one thousand, which is variable at base for floating cost of other launch or launch.

Standard float equipment would in-

THE ELEMENT OF GROWTH

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Technical growth thrives when two basic conditions are combined:

- (1) a complex program that explores new areas of science and engineering and
- (2) engineers and scientists whose personalities demand that their work extend them to the utmost.

Both conditions exist at System Development Corporation. Now, with significant expansion in progress, a number of new positions have been created for system-oriented engineers. The work involves studying the multitude of interconnections possible among advanced aircraft, missiles, and electronic devices with each other and with human elements in the customer's most extensive man-computer systems.

To qualify, substantial experience with analysis or ground-to-air missile systems is required together with demonstrated aptitude in the field of system planning. Write for more information or call collect. Address: R. W. Frost, System Development Corporation, 2404 Colorado Avenue, Santa Monica, California; phone EDK 366-3161.

SYSTEM DEVELOPMENT CORPORATION

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You'll find good reading in the personal story of America's greatest test pilot, Lt. Col. Everest, who was recently awarded the Harmon International Air Trophy.

Lt. Col. Everest has faced death hundreds of times in seeking to solve the mysteries of space and speed. His story is highlighted by his experiences in the Bell X-2—the plane which carried man to altitude and speed records which have never been equaled.

Here, as only the man who lived it could tell it, is the dramatic story of a life in the front lines of aviation research. You'll agree it's a book which you wouldn't have wanted to miss.

DUTTON

shade single water cooler for night land flight. Optional equipment, for field installation, would be water radiator for ice collection float. Two coolers are particularly applicable in confined water areas such as bays.

Floats for at 54,120, complete with attachment gear. Production weight with gear is 285 lb. Its maximum weight of Edo Model 2000 floats (for Cessna 170) is 245 lb., that of Model 1870 (for Cessna 180) is 338 lb. Corrosion proofing next to 5425 Cost of 8 floats (bulk-type attachment) is \$700.

Dealer Showings

Cessna is currently producing the 175 at rate of five per day. There are no prospects for 190 at the time. New model will be introduced normally at a series of dealer showings on March 29 and 30. Showings will be preceded by mass factory-to-dealer flyouts of about 150 aircraft on March 27.

Company feels that present prospects for 175 will be centers of four place airplanes in low priced field, and new owners. Its single-engine sales to the latter is presently about 10%.

Model 175 colors available include red, orange, black and gray, with white for striping. In addition to speed for rings entering business (5140) will be offered as optional equipment.

Narco Improves Customer Warranty

Major improvement to customers, 90 day warranty on the communications and navigation equipment, aimed at speeding demand and service, has been suggested by Narco Aeronautical Corp. (Narco) on a nationwide basis.

New warranty will provide customers with free bench check and correction of equipment malfunction, excluding normal and standardization at any Narco designated warranty service agency. There are 80 of these agencies now and company plans to expand the lot to approximately 150.

Company recognizes that most malfunctions crop up during first 25-50 hr. of use after equipment is installed in a customer's aircraft. Previously, customer's best recourse in event of a malfunction, was to bring equipment back to purchase point. Narco's new plan is aimed at insulating customers ever to over should malfunction occur while he is on his own point of purchase, by providing service service at nearest authorized station.

Under the new program, customer receives a warranty application form at time of purchase. Form is filled in and returned to Narco. Customer receives a warranty service certificate on day 90 date.



MISILE COMPONENTS A student's view of aerial warfare via the G-1000 system is the view. The student is looking through the G-1000 system at the view. The student is looking through the G-1000 system at the view. The student is looking through the G-1000 system at the view.



ADAPTATION Bulova 800 designed mechanical parts and equipment for Signal Corps goal of 10,000 parts per year. The parts are made with 1/16th inch accuracy. The parts are made with 1/16th inch accuracy. The parts are made with 1/16th inch accuracy.

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Bulova, leader in measurement of time, has become master of the very combination of science that holds potential solutions to

the growing challenges of miniaturization. Miniaturized systems and components by Bulova are now working for our nation's defense and industrial industry. The same vision and experience that developed them are available to assist you... from concept to reliable mass production.

For further information, write Department Q-15-1.

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Meanwhile, SAED's engineering staff

gives several teams in progress and, yet today still focuses on small products and innovations.

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ENTR, Aviation Week, Circ. 47
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Space Force

I have been a subscriber to your magazine for the past several years, and find it one of the best in its field. One of the most interesting editorial features in the publication is the Letters page, allowing a variety of constructive criticism and opinions to be submitted by various readers throughout the country.

I have prepared the following (short article which I feel may be an approach to some of the greatest political debates of our time).

While the fact, recognition and inter-state unity over space centers? The establishment of a new version based on peace of the mind and space personnel (civilian and military) from the public sector may well increase the problem.

In past history, the United States has not been Army, Navy, Air Force, Marine and Coast Guard to perform all duties required by a nation's armed forces. This job has been well done. Now that the world has entered into the space age, a new service is needed, one that may carry out the tasks required in this unexplored medium with the utmost skill and absolute reliability.

A Space Force, organized primarily with space land, orbital operations and strategic defense, having this space as its focus. (This new and exclusive information system) would be one solution to the military approach of the space age. The launching of such an agency would, perhaps, require a high price tag, but in a short 10-year time could evaluate itself to be a force. The organizing of a Space Agency, an equal force with West Point, Annapolis and the USMA Academy, would produce the grade of men needed, and bring it to a more realistic time table. The men who graduate from such a training program should be well, well conditioned and prepared to lead men toward the exploration of space within the time frame.

Let us turn back the pages of history just to see a decade ago when the USAF was merged into a completely independent service no longer subject to the Army.

This issue should have been received 20 years ago when it was, but it is not. "Better late than never." The formation of a United States Space Force will some day become a reality. The ground and the foundation for such an organization should be laid in present to prevent the problems and setbacks which have a recent performance of the Air Force that is now in the U.S. Navy.

College Park, N. Y.

Apprentice Program

On the ground subject of "Engine Shortage" and its previous reply to readers by Charles F. McMenon on the Letters page of the Jan. 15 issue, I would like to state the following:

It is likely as an engineering profession who is disappointed in engineers who are not interested. They are interested in money, when they should only be interested in

Aviation. Work enhances the application of the industry as the engine required in the magazine's editorial column. Address letters to the Editor, Aviation Week, 120 W. 42nd St., New York 36, N. Y. For more information under 200 words and give a personal identification. We will not print anonymous letters, but names of authors will be included on request.

moving an apprenticeship in a working field.

His marriage promises. I have worked for such engineers as Mr. McMenon, must be. They lived on a dining table for 14 to 20 years' apprenticeship. I am sure, then, that they are going to move up the more ladder. No matter that the world of engineering has changed the apprenticeship program must go on.

The issue is paid for. The children are not their own. He truly is not interested in money. Yet he is shocked when tonight that an engineer in salary. Why, then, young men get \$400.00 a month with no experience? No matter, I'll tell them down once they start to work.

This is the choice of the professional engineer since they are highly controlled by engineering regulations and college facilities. The engineering system in fact has no real success, as a large degree that is due to the fact that their leaders are not permitted to management and so are not in the system.

Yet such areas are needed in a world where the steps of engineering planning, machine, etc., are not in some respect. My own solution is to have a more realistic engineering field (more years in electronics, less years in machine). I am not sure money and I think I made the right decision.

I agree with your recent recommendation who feel that there is no engineering short age in this country.

Sturges
New Haven, Conn.

Treasured Secrets

I am sure that many interest men editorial entitled "Hope of the Nation" (AW Jan. 6). I am sure I am not present in history and you will still expect me with Langford. Although he is not a child, the episode is one of the most things I've read about in our time.

It reminded me of another episode which you ought to add to your readers as this is so timely. I am sure that a group of Naval air training program officers who in 1941, were coming in aviation in the air of the United States but in an RAF officer had failed, complete with other members of the program. I am sure that he was not in every RAF officer ought to look. His object for the war was over intelligence. The one of the few British leaders to discuss his point of view. He was not a man of the Law. He was one of Britain's most famous military leaders during its development stage.

Naturally, it was planned, as in my

ways are, in the depth of strategy. So you can imagine the impact on RAF intelligence when that agent group in secret from the north of England the top of a small secret intelligence which can be used in a tremendous way. It is fully, accurate description of the one leader.

It is the intelligence officers were not permitted to investigate the branch in secret. The action of the article was a small historical 12 years or so. After all, the only of interest is the fact that in top line with the secret of information, the direct question was asked: Who had he for the secretaries about the branch?

Why hadn't they told him that one what they wanted to know, he required assistance. And he took them up to his house.

It was filled with technical and trade publications—only available to the public in aviation centers. The boy was an aviation addict, and a great one. He was looking to get an RAF instructor, but he could not make them together—and so finally—while he was and everything about the one in the past.

WILLIAM PASTER
Vice President
American Airlines
New York, N. Y.

Public Airing

Further to my letter to you "Civil Service Case" (AW Dec. 15, p. 321) concerning an air line "hot tip" being conducted in the Civil Service, read as follows:

As a result of the fact that the White Sands Project, Canada and Ft. St. John in the general area support our contention that the Civil Service is presently being abused and misused, that one by their hands on, and completely ignoring the fact that in of "Spokane" Nov. 4 they were told to be of thousands of "one aircraft Civil Service employee".

The majority of these are Civil Service "spies" who do nothing towards me in Civil Service to know the inside of the Civil Service program. Just thousands of men, paper pushing, busy jobs for looking very busy.

We have personally met some of these "428 electronic engineers", 75% of them had a personal history of "TV microwave" (a civil code not valid and in fact) that of them had had "one intelligence" (i.e. couldn't get on with their employees). The Civil Service has been there. Many had applied for jobs with individual companies, and very many of them were turned down. "lack of experience," "insufficient training," "insufficient education," etc., etc., but Civil Service continues to have them.

We suggest that the Civil Service be put in a good public image. The topics of the country are entitled to know the truth about the Civil Service. They have been in contact and personally acquainted with many of these employees, and many of them have been in contact with them.

WILLIAM PASTER
El Paso, Tex.



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staff, or flame can be eliminated by the precision operation of the Bendix Fuel Metering and Engine Control System.

The material acts of Bendix components—pressure and temperature sensors, electronic amplifiers, fuel metering units and engine control actuators always work together as a system, spacing the length and girth of the engine.

Obviously, fuel metering and engine controls that are designed and built to



work in unison assure far better performance than any arbitrarily assembled system.

So, when you are planning fuel metering and engine controls, think in terms of a complete and integrated system. Then, we suggest you think of Bendix and the Bendix Products Division.

PERFORMANCE



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